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NAVAL POSTGRADUATE SCHOOL Monterey, California



THESIS

MULTIPLE TARGET IDENTIFICATION

AND DIRECTION FINDING

USING MATCHED FILTERING TECHNIQUES

Ъу

James L. Johnston

December 1983

Thesis Advisor:

H. A. Titus

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T21-216



REPORT DOCUMENTATION	ON PAGE	READ INSTRUCTIONS BEFORE COMPLETING FORM
. REPORT NUMBER	2. GOVT ACCESSION NO.	3. RECIPIENT'S CATALOG NUMBER
Multiple Target Identification Finding Using Matched Filterin	5. TYPE OF REPORT & PERIOD COVERED Master's Thesis December 1983	
11.101.15 001.15 1.010.10 1111011.	.8 10mm4400	6. PERFORMING ORG. REPORT NUMBER
. AUTHOR(e)		8. CONTRACT OR GRANT NUMBER(*)
James L. Johnston		
PERFORMING ORGANIZATION NAME AND ADDR Naval Postgraduate School Monterey, California 93943	ress	10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS
- CONTROLLING OFFICE NAME AND ADDRESS		12. REPORT DATE
Naval Postgraduate School		December 1983
Monterey, California 93943		13. NUMBER OF PAGES
. MONITORING AGENCY NAME & ADDRESS(II dill	erent from Controlling Office)	15. SECURITY CLASS. (of this report)
		15a. DECLASSIFICATION/DOWNGRADING

Approved for public release; distribution unlimited.

17. DISTRIBUTION STATEMENT (of the obstrect entered in Block 20, if different from Report)

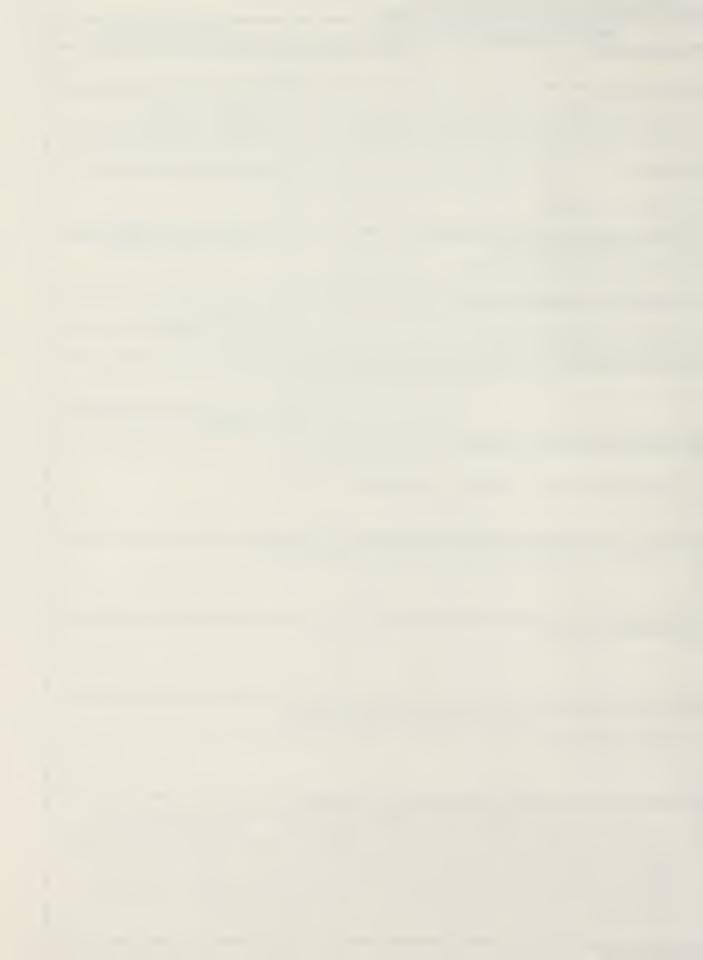
18. SUPPLEMENTARY NOTES

19. KEY WORDS (Continue on reverse side if necessary and identify by block number)

seismic sensors, matched filtering, target identification, target direction.

20. ABSTRACT (Continue on reverse side if necessary and identify by block number)

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Multiple Target Identification and Direction Finding Using Matched Filtering Techniques

b y

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Submitted in partial fulfillment of the requirements for the degree of

MASTER OF SCIENCE IN ELECTRICAL ENGINEERING

from the

NAVAL POSTGRADUATE SCHOOL December 1983



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ABSTRACT

This research investigates seismic signal processing techniques for battlefield target classification and acquisition. Multiple target classification is performed by discrete time domain matched filtering. Multiple target directions are determined using the responses of the matched filters and least mean squares polynomial curve fitting. The least mean squares polynomial curve fitting procedure is also used for direction finding for recoil/blast sources, using the unfiltered seismic signals.



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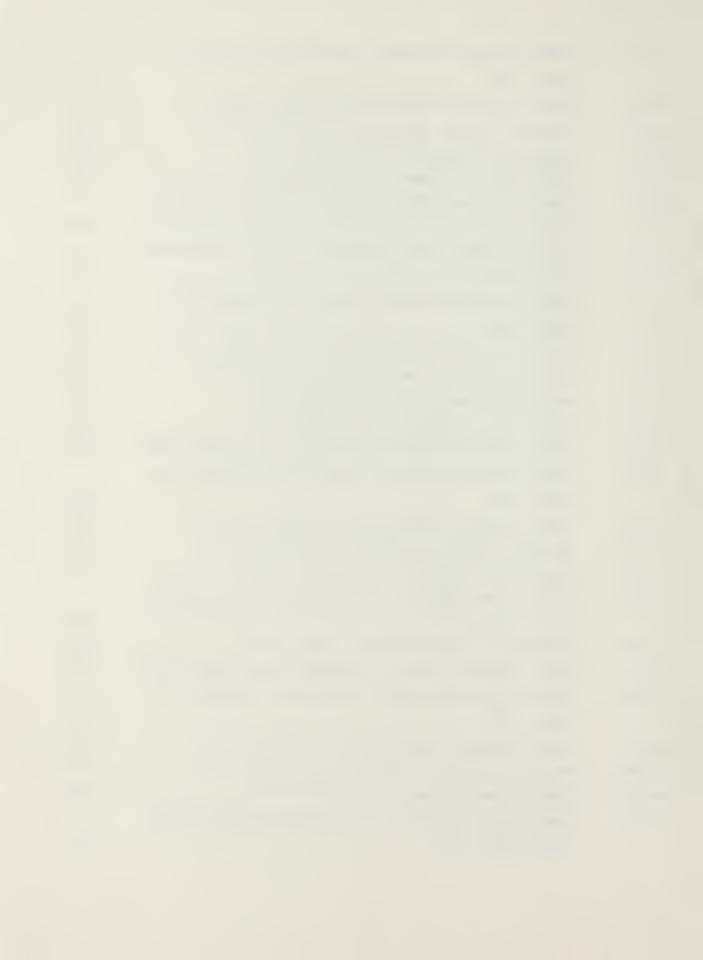


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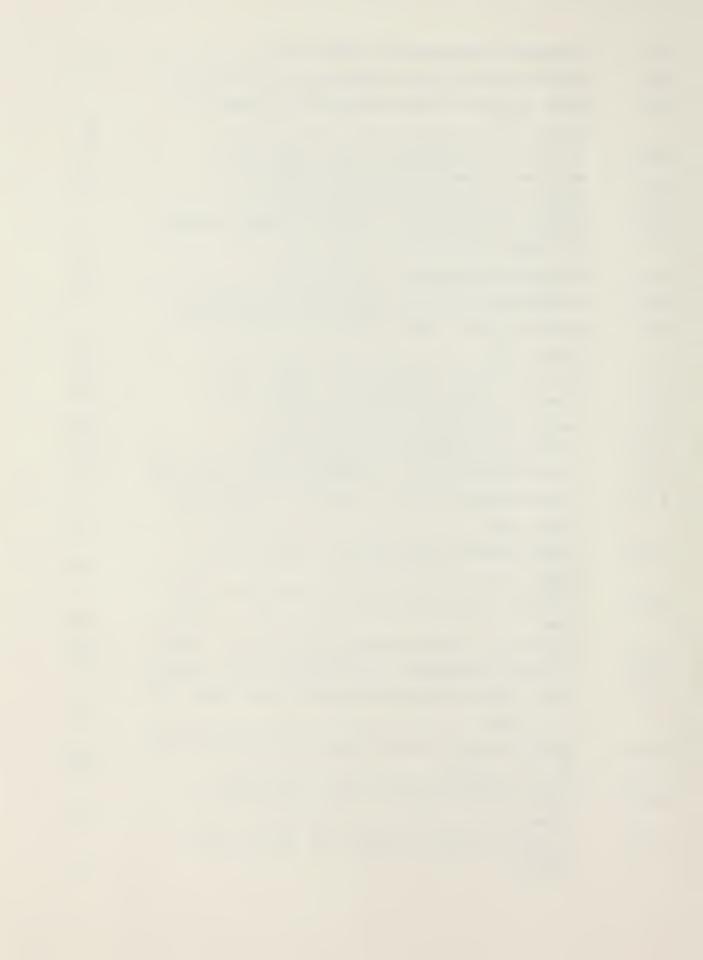
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ACKNOWLEDGEMENT

I wish to express my appreciation to Professor Harold A. Titus for his professional guidance and support in developing this thesis. I would also like to thank my wife, Pamela, for her constant support during the writing of this work.



I. INTRODUCTION

part of the modern battlefield. A primary goal of combat intelligence is target acquisition. Hostile targets can be acquired by either passive or active means. An example of passive target acquisition is visual target identification. Radar, on the other hand, is an active target acquisition device. An effective combat intelligence system will include a mix of both active and passive target acquisition methods.

Rapidly advancing technology in the fields of electronic counter-measures and radiation-seeking weapons has enhanced interest in passive target acquisition methods. To be cost effective, as an additional target acquisition system, a passive system must be able to provide swift and accurate target identification, location and tracking information on hostile targets. A variety of seismic sensor systems have been used in this roll with varying degrees of sucess.

Naval Ocean System Command (NOSC) in San Diego has developed a system based on a circular ring of sensors with data collection managed by an array-processor/minicomputer system [Ref. 1]. The observation of enemy movements and activity beyond the Forward Edge of the Battle Area (FEBA) is the design objective of this system. This thesis uses data collected during a test of this system at the Marine Corps Air-Ground Combat Center at Twenty-nine Palms, Califiornia. Investigated are various methods of processing the seismic data collected. The objective of this research is to try to provide viable methods of satisfying system design objectives through signal processing techniques.



The following chapter addresses the design objectives and requirements for such a seismic sensor system. Additionally, capabilities and deficiencies of current systems and research are detailed. In order to intelligently address solutions to these requirements and deficiencies, an understanding of seismic theory and sensors is needed. General seismic theory is presented with emphasis on the constraining parameters for the use of the earth's surface as a medium for gathering seismic intelligence. Also highlighted are the similarities of the earth's surface to electro/optical phenomena and the resulting simplifying assumptions.

The sequence of design solutions investigated followed from the analogies and simplifying assumptions addressed in the study of seismic sensor theory. The problem of target identification or classification is approached using digital matched filtering of the time domain amplitude data. Frequency domain matched filtering was not considered, based upon the conclusion by NOSC that there appeared to be no consistent spectral lines for any of the possible target types, except for artillery [Ref. 1]. Matched filtering was used to identify single and multiple target classes occuring during a sample period. The chapter detailing the matched filter procedures and implementation also includes a description of support and validiation software used in the analysis of the seismic data.

The validation software is primarily used to check the accuracy of the direction finding routines. These direction finding routines are the time domain phase difference procedure (TDPD) and a least mean squares polynomial (LMSP) curve fitting procedure. The combining of the matched filtering procedure and these direction routines allowed for multiple target direction finding. The theory and derivations of the two direction finding algorithms are presented in the multiple target direction chapter.



Application of these algorithms is performed first on simulated targets for validation of the procedures. The experimental data is then analyzed. A user's manual, which includes procedures for tape and mass storage operations, is provided as an appendix. This appendix describes how to set up and use the software system.



II. THEORY OF SEISMIC SENSORS

Elastic waves result from the stressing of an elastic media. The elastic media for seismic theory is the earth. Seismic theory is the study of the earth as a wave propagating media. Elastic waves propagate away from the source of seismic stress, e.g. an explosion [Ref. 2]. The energy, which propagates through the earth, travels via particle deformations. The elastic properties and densities of the earth media determine the velocities of these seismic waves. [Ref. 3]. Seismic wave sources of interest may be impulsive or continuous. Impulsive or short duration sources artillery recoil or shell blasts. The time-limited nature of this type of seismic signal produces a broad range of frequencies. Continuous wave signals may be produced by tanks, trucks and low flying aircraft. These continuous wave signals may be described by narrow band frequency characteristics. The spectral power of a seismic source is a function of several parameters. A non-inclusive list of these parameters includes:

- 1. The vehicle's velocity and mass
- 2. The size of the explosive charge associated with the artillery or shell blast
- 3. The degree of coupling into the earth's surface
- 4. The geological structure over the wave's path

Information about the seismic source is contained in the waves which it generates. For example, in an array of seismic sensors, directional information is contained in the relative received signal phases. In otherwords, the relative phase differences between the signals received by the individual array elements can be used to compute the direction to the seismic source [Ref. 4]. These relative phase



differences represent the time delays of the waves as they pass the array's elements. The response of the array's elements is proportional to the amplitude and velocity of the earth's motion, relative to the geophone's sensing axis for the waves. [Ref. 3]

Assumptions about the propagation of seismic waves must be made to assist with their analysis. The earth is assumed to be made up of horizontal, homogeneous, and isotropic layers of material. These layers are assumed to be discontinuous in their elastic properties at their borders. This variance in the elasticity between the layers leads to an optical analogy for the wave propagation. Propagation paths may now be viewed as being direct, refracted, or reflected versions of the source's seismic waves. [Ref. 3]

There are four basic types of seismic waves. These types are compressional, shear, Love, and Rayleigh waves. Compressional waves are generated by impulsive sources such as shell blasts. Particle motion is along the direction of travel. Shear waves are characterized by particle motion orthogonal to the direction of motion. The Love wave is a surface wave which may occur as a result of the layering of the earth's surface. This layering effect acts as a wave guide for this type of wave. Particle motion is orthogonal to the direction of wave propagation. The Rayleigh wave is generally the strongest of the seismic waves. The Rayleigh wave travels along the free surface of the earth. Its particle motion direction is always in the vertical direction. It is the strongest wave generated by a compressional source. Its amplitude attenuates at a rate only inversly proportional to the square root of the distance. [Ref. 3]

The waves of primary interest for a seismic system are the Rayleigh and Love waves. This is due to the long-range propagation of these waves. Since these two wave types are orthogonal to each other, they must be sensed by different



geophones. Rayleigh waves may be sensed by vertical geophones and the Love waves by horizontal geophones. Since the Rayleigh wave is normally the strongest, and in order to reduce computational complexity, only verticle sensor data is used.

Rayleigh waves experience absorbtion losses, particularly at higher frequencies. This phenomenon occurs because of the lowpass filter effect of the earth. This filtering effect is further compounded due to the fact that the cutoff frequency of the earth diminishes with range. Further complications arise due to the dependence of wave velocity on frequency. The result being that the wave train may change with distance, reducing the correlation of the wave shape between its source and distant points. Other sources of error occur because of the weathering of the surface layer, irregularities in the sub-surface composition, variances in the earth's layers and surface geometry. [Ref. 3]



III. DEVELOPMENTAL REQUIREMENTS AND PROBLEM DEFINITION

A. PASSIVE TARGET ACQUISITION AND SURVEILLANCE

A battlefield commander possesses a definite requirement for real time combat intelligence. A significant tactical advantage is held by the commander who is able to integrate his available combat intelligence sources with his supporting arms, i.e., target acquisition and engagement. It follows that to be part of the target acquisition process, any real time, seismic sensing system must provide swift and accurate information on detected enemy targets. The specific requirements for such a system are the ability to detect, identify, and locate these targets [Ref. 5]. Additionally, the target's rate and direction of movement should be provided or made easily discernible.

Any seismic sensing system must be designed around the target acquisistion cycle. The target acquisition cycle, as given by Dublin [Ref. 5], is as follows:

- 1. Search Time
- 2. Target Sensing
- 3. Information Processing
- 4. Display of Target Information
- 5. Analysis of Target Information
- 6. Time required to make a Decision
- 7. Time Required for Supporting Weapons to Respond

For a seismic system, a prioritized list of possible targets are as follows:

- 1. Artillery
- 2. Helicopters and Aircraft
- 3. Tracked and Wheeled Vehicles
- 4. Personnel



As may be expected, the relative amplitudes of these seismic targets vary widely. A seismic targeting system is therefore constrained as to the targets it can or can not be expected to effectively engage.

The variance in the relative amplitudes of seismic targets suggests a range of specifications for detection of these targets. As summarized by Dublin [Ref. 5], possible detection radii may be as shown in Table I. Radii are given for both short and long targeting systems.

TABLE I Target Detection Radii									
Target	Short	Range	System	Long	Range System				
Personnel Vehicles Low Flying Airo Hostile Weapons	craft	•	100 M 1 KM 1 KM 1 KM		Non = 10-20KM 10-20KM 15-20KM				

The timeliness requirement is ancillary to the radii of detection specifications. Timeliness, as used here, refers to the total time commencing when the seismic sensor system first detects and processes the seismic target data and ending with the dissemination of the targeting information to command elements for disposition. This timeliness requirement ranges between five to fifteen minutes, depending upon the mobility of the target [Ref. 5].

The parameters having a direct effect on the timeliness of a system are the probabilities of false alarm and detection. These parameters directly relate to a system's value. Increased probability of detection with reasonable false alarm performance, combined with the ability to disreguard



friendly targets, are practical design objectives for any targeting system. The ability to incorporate such design features into a seismic sensor system will reduce both the time wasted on invalid targets and the danger of undetected targets.

Once a valid target has been detected, target location information must be obtained. Stringent specifications for target locations allow for the system to support or enhance the effectiveness of; fire support systems, blind bombing, Harassing and Interdiction fire (H and I), and observerless artillery engagement.

B. CURRENT CAPABILITIES AND DEFICIENCIES

To date, numerous successful algorithms have been developed to determine direction to single targets. These algorithms include both time and frequency domain methods. Target identification of long range targets via seismic sensing has not-yet met with equal success.

The modern battlefield is seldom a single target type environment. The complicated, real world problems of multiple target identification and multiple target engagement require solution before practical seismic sensor systems can be integrated into the target acquisition process.



IV. MATCHED FILTER CONCEPT AND DISCRETE ALGORITHM

A. MATCHED FILTER THEORY

As previously addressed, there exists a requirement for battlefield target identification/classification. The recovery and classification of target signals suggests a filtering requirement. Previous works and implementations have used frequency domain techniques [Ref. 4]. The Air Force's SKEET system and the U.S Army's Remote Battlefield Surveillance System (REMBASS) both have successfully implemented a spectral power approach for classification of seismic data. These systems, however, are for short range applications. Time domain approaches to target identification have been for the most part left unexplored.

The discrete matched filtering technique is an attempt to classify targets by their time domain amplitude pattern i.e., their seismic amplitude signature. The heuristic basis for this method evolved from the observation of visual differences between the amplitude versus time signals for the various classes of targets. The matched filter, being the optimum filter for detecting known signals, was selected [Ref. 6].

The discrete matched filter is described by equation 4.1, where h(t) is the impulse response of a filter whose output signal to noise ratio at time t_0 is maximized. The unit step u(t) has been added to assure causality for the system. The matched filter is [Ref. 7].

$$h(t) = s(t_0 - t)u(t)$$
 (4.1)



The output signal is given by

$$sout = \int_{-\infty}^{\infty} h(v) s(t - v) dv$$
 (4.2)

The maximum signal to noise ratio is given by

$$(\text{sout}^2/\text{M}^2) \text{ maxoutput} = E(t_0)/N_0$$
 (4.3)

where M is the noise level at the filter output, N is the input noise level, and $E(t_o)$ is the energy in s(t) up to time t_o . In equation 4.2, the replica of the original known signal is reversed and translated in time to be convolved with the signal input to the filter, producing the optimum output signal to noise ratio.

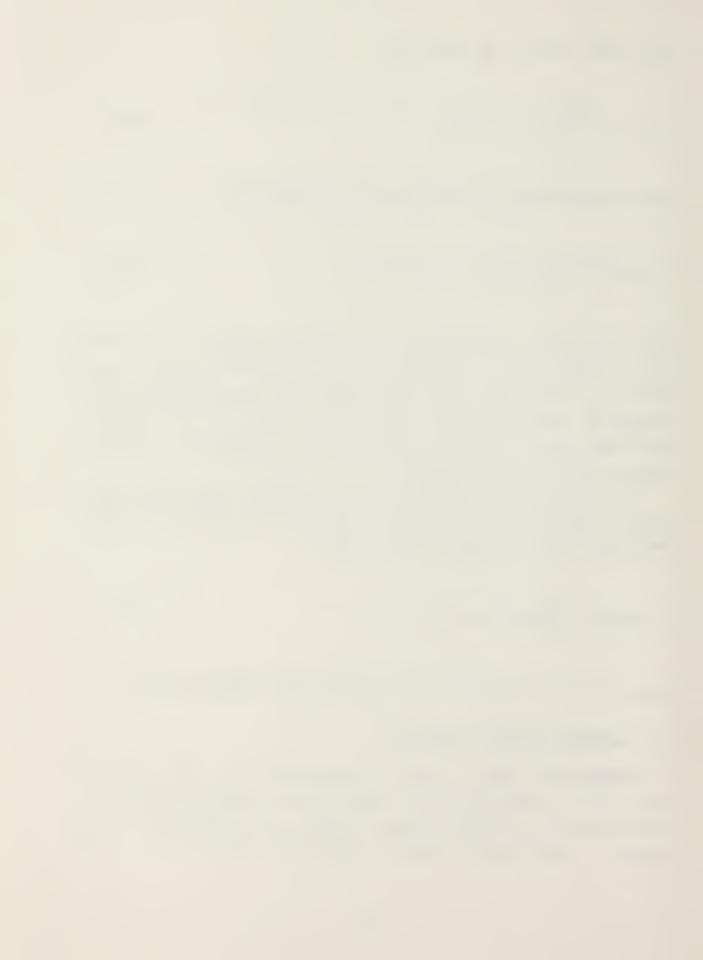
For discrete realization of the matched filter as implemented, equation 4.2 becomes equation 4.4, where h(k) is the reflected and translated known signal.

sout(j) =
$$\sum_{k=1}^{N} h(k)s(j-k)$$
 (4.4)

Where N is the number of data points per sample period.

B. MATCHED FILTER ALGORITHM

Samples of known signals are stored in a data file and read into a 5120 array at the start of program execution. Five sample, or known filter signals, are recorded in this array. Each sample signal comprises 1024 of the 5120



elements of the matched filter array. To perform the matched filtering, the 1024 seismic data samples (unknown signal) are copied over five times on an array of 11264 size. These copies of the experimental data are separated by 1024 zeros on either side. Additionally, the leading and trailing elements of the experimental data are set to zero to eliminate switching spikes present in the data. Hence forth, this 11264 element array will be referred to as the working array for brevity. This working array will contain the results of the matched filtering. For M, given as the number of input signal and filter signal array elements and also the number of zeros, the requirement set forth in [Ref. 8] for nonoverlapping convolution of length L is satisfied. Equation 4.5 establishes the minimum length for nonoverlapping convolution of one matched filter segment.

$$L = 2M - 1 \tag{4.5}$$

Figure 4.1 depicts the working array layout. Notice that the first filter signal, "Tracked Vehicle" is loaded into the h(t) array and is convoluted with the working array elements one through 2048. Once the leading data point of h(t) reaches the working array's element 2048, a new known signal is loaded into h(t), "Wheeled Vehicle", and the convolution is continued for working array elements 2048 to 4096. This process continues until the last of the five h(t) filter signals has been read in and the convolution has been performed on all copies of the data. Notice that working data array elements 10240 to 11264 are for array length over-run protection.

Prior to convolution, each filter sample signal and the input signal are equalized to the same power level through



1024 204		ZEROSDATA ZEROSDATA ZE	
VEHICLE	WHEELED SHELL VEHICLE BLAST	HELO PERSONNEL	
* - Indica is loa	ates point at which aded into h(t).	new known sample signa	1

Figure 4.1 Working Array Configuration

division by their respective root mean square values. Additionally, after the convolution, the entire array is normalized with respect to its maximum amplitude element. The maximum value in each target classification section is then compared with the interactively selected matched filter threshold. If the section's peak value is above the threshold value, that class of target is declared to be present. This method allows for the simultaneous detection and classificaton of multiple targets. It should be noted that this equalization forces both high and low amplitude signals to an equivalent average power This is felt to be justified in that signal amplification is not the goal, rather target detection and classification is. In the case of a single target, the known signal that most closely matches the unknown signal is anticipated to have the greatest amplitude matched filter spike. Figure 4.2 is a sample output.



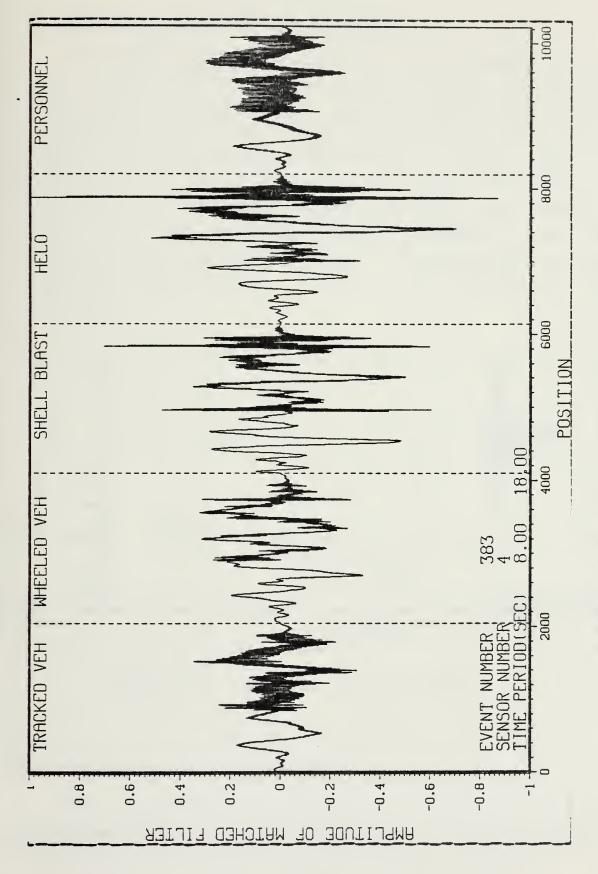


Figure 4.2 Sample Matched Filter Output



C. MULTIPLE TARGET MATCHED FILTERING

The upper limit of equation 4.4 may be selected to be from one to 1024 when called by the multiple target direction routine. This allows for the selection of reduced program execution times. Target identification, however, is always made by the full 1024 element buffer. When the matched filter target identification routine is used by the multiple target direction finding routine, data windows of less than 1024 are formed from a segment of the 1024 elements by extracting the segment size required around the maximum signal value. Figure 4.3 illustrates this procedure.

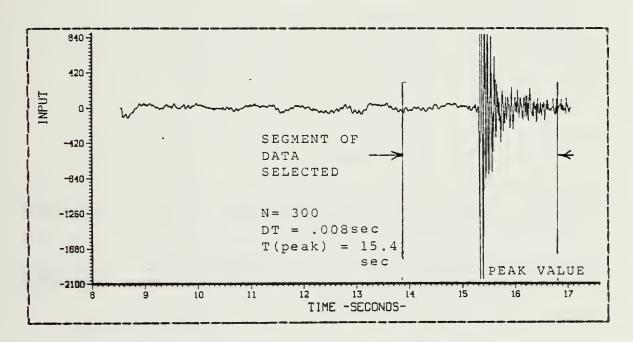


Figure 4.3 Windowing of Experimental Data

D. COMPUTATIONAL REQUIREMENTS

The computational requirements for this procedure are described by equation 4.6, where NO is the total number of operations required by the matched filter routine.



Ndf is the number of data elements and also the number of filter elements. No is the total number of target classes. The term Ndf2Nc is the total number of multiplications required, while (Ndf - 1) NdfNc is the total number of additions.

When used for target identification, Ndf equals 1024 and Nc equals five. Equation 4.6 gives a total of 10,480,640 operations for these array sizes. When the matched filter routine is initiated by the multiple target direction routine, the value of Ndf can be selected to range from one to 1024. Figure 4.4 is a plot of equation 4.6 and shows the computational consequence for selection of large values for Ndf. Note that equation 4.6 must be multipled by the number of sensors in the ring when computing the number of operations for the multiple direction routine.

Window sizes of 100 and 200 were found to provide acceptable accuracy with greatly reduced computation times. As shown by figure 4.4, a window size of 200 requires 399,000 operations, while a window size of 100 requires only 99,500 operations. The number of operations required was found, as expected, to be proportional to the execution time of this routine.

E. SUPPORT SOFTWARE

1. Amplitude Analysis (Timout)

A graphical output is provided by this amplitude routine which displays the relative amplitude versus time for a selected sensor. The initial target direction and target classifications found are also displayed. This routine allows for the interactive selection of the



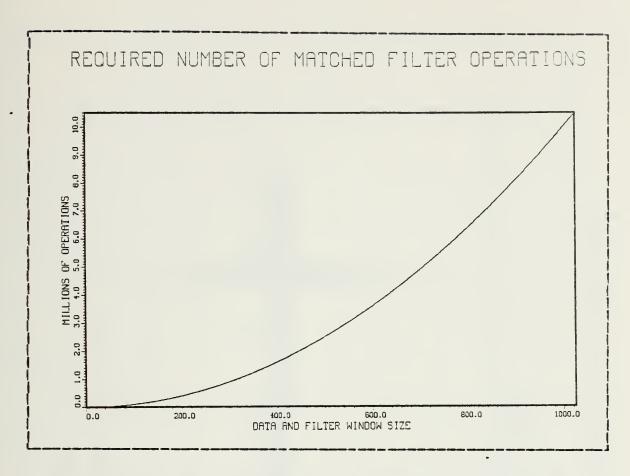


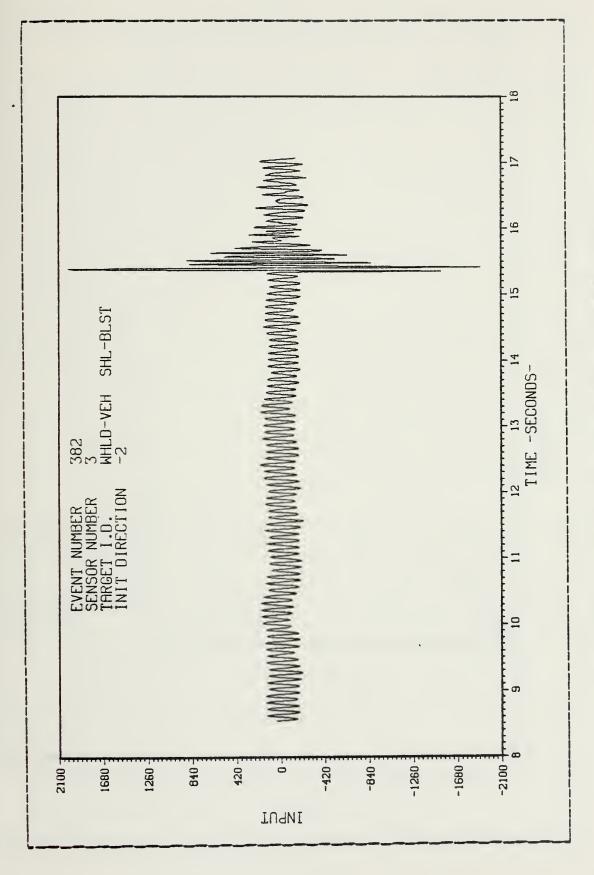
Figure 4.4 Data Window Size vs Number of Operations

amplitude response of any sensor as a sample target for later use in the matched filter analysis. In this way sample, signals can be catalogued and evaluated as filter signals. The axes of the graphical output adapts to the data's maximum amplitude and to the time period involved. Figure 4.5 is an example of amplitude analysis output.

Frequency Analysis (Freqot)

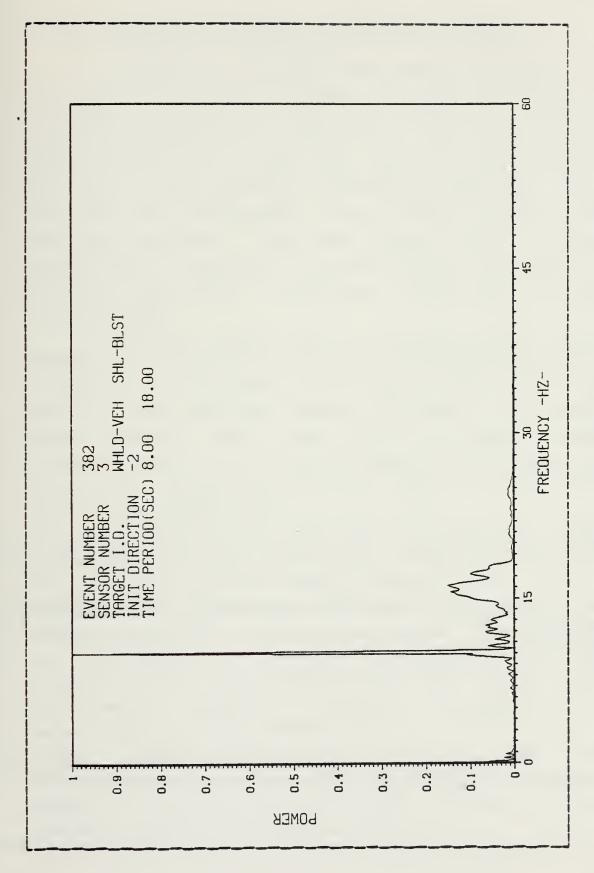
The frequency routine, as in the amplitude analysis routine, allows for initial primary target direction, target classifications, and for the time period of the experimental data to be displayed. Normalized spectral power versus frequency is graphically displayed as shown in figure 4.6.





Sample Amplitude versus Time Output Figure 4.5





Sample Prequency versus Power Output Pigure 4.6



3. Simulation and Validation Software (SIMULT)

In order to validate the various algorithms and their implementation in software, a testing procedure was required. The specific algorithms which the simulation routine was designed to validate are the initial angle, phase difference and least mean square target direction routines. These routines, as previously described, use the relative time differences of the seismic signal's peak amplitude response or the peak matched filter response respectively. Validation of these routines is performed by allowing the creation of simulated targets with selected arrival angles.

Simulated targets can be created in the routine SIMULT. Up to four sine wave targets of selected frequency, amplitude, and direction can be input during each sample period. These simulated targets are added to the experimental seismic signal data for the sample period. Correspondingly, zero direction filter data must have been written into the matched filter data file at these selected sample target frequencies.

The directions for the simulated targets are created by introducing relative phase delays between the sine waves that are added to each sensor's seismic data. This is implemented by introducing a zero phase to the sensor in the desired direction of the simulated target. The phases of the other sensors are increased proportionally by their distance on the circular array away from the zero phase shift sensor. Figure 4.7 illustrates the case of a zero degree simulated target direction.

A summary of the target directions found by the multiple target direction routine and the simulated targets entered is provided by the multiple direction plotting routine. Figure 4.8 is a sample output.



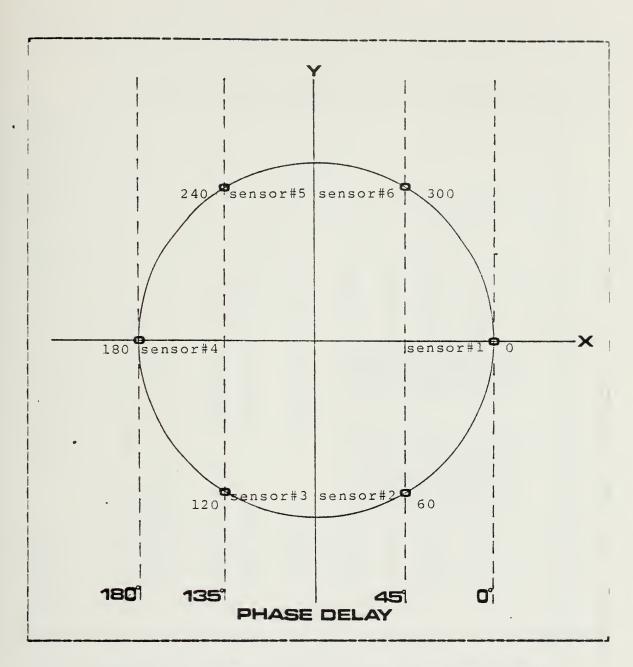


Figure 4.7 Simulation of a Zero Degree Target



MULTIPLE TARGET - MATCHED FILTER OUTPUT

			00.00	00.00	0.00	00.00	
375) 25.00 35.00	DIRECTION59.00 DIRECTION - 315.00	DIRECTION59.00	SIMULATED TRKD VEHICLE TARGET FREDUENCY AMPLITUDE 0.0000	ON 0.0000 KEHICLE TARGET FREQUENCY DE 0.0000	ON 0.0000 JPTER TARGET FREQUENCY DE 0.0000	0N 0.0000 INEL TARGET FREQUENCY DE 0.0000 DN 0.0000	
EVENT NUMBER 375 TIME PERIOD(SEC) 25.00 35.00	MIEELED VEHIGLE SHELL BLAST	PERSONNEL	SIMULATED TRKD V	DIRECTION SIMULATED WHLD VAILED VAILE	DIRECTION SIMULATED HELICOPTER 1 AMPLITUDE	DIRECTION SIMULATED PERSONNEI AMPLITUDE DIRECTION	

Sample Multiple Target and Simulated Target Output Figure 4.8



V. MULTIPLE TARGET DIRECTION

A. THEORY AND DESIGN CRITERION

The modern battlefield is comprised of many classes and quantities of seismic signals. Using these signals for target identification and acquisition is the purpose of battlefield seismic sensors. As presented in the last chapter, matched filters can be used for target identification. In this chapter, it will be shown that matched filter information may also be used to obtain target bearing.

Multiple target acquisition requires the concurrent separation of the target classes and the computation of direction for each of the target classes found. The output of the matched filter is a spike at to, the time of peak signal detection. If matched filtering is performed for each of the sensors in the ring, the value of their respective to, for each filter output, would be different. Since the seismic waves impinge upon each sensor at different times, dependent upon the target direction, the values for to for each sensor may be expected to be directly related to the arrival angle of the seismic wave. The to spike of the matched filter output may be thought of as a signal compression for both the continuous (tank, truck etc.) and the time limited (shell blast, artillery recoil etc.) seismic signals [Ref. 6].

Time domain methods [Ref. 1] use the positional differences of known wave points to geometrically estimate direction. The times associated with a sensor ring's peak amplitude responses may be explained by way of an illustrative example of a shell blast. A rough direction to the origin of this shell blast can be computed using the



relative time differences associated with the peak amplitudes of all of the ring's sensors [Ref. 9].

The enhanced time positional information, which is a by-product of the matched filtering, can be used to perform just such a time domain approach. The motivation for the method is that, unlike other time and frequency domain methods, which are very susceptible to noise corruption inaccuracies, matched filtering pulls the signal out of the noise and optimally detects the signal at time to. Two time domain methods are evaluated for finding arrival angles of the seismic signal. The first being the time domain phase difference (TDPD) method. The second being a least mean squares polynomial (LMSP) curve fitting approach.

B. MULTIPLE TARGET FILTERING ALGORITHM (MULTI)

The routine MULTI calls the matched filter routine for each sensor's amplitude signal. Returned are the target classes found with their relative peak filter response positions. Figure 5.1 illustrates a two target case. matched filter response and the relative time postion for the two classes of targets can be seen. The figure shows that a shell blast target is present. The relative time for this target class is 5800. The relative time returned for the simulated wheeled vehicle target is 3000. These times or positions are relative since each sensor's filter response peaks are offset in time with respect to the peaks of the other sensors. This allows for simultaneous direction finding for each class of target. This algorithm allows selection of either a time domain phase difference or or a least means square polynomial algorithm for finding the direction to the targets. The time domain phase difference algorithm is derived first [Ref. 1]. The least mean squares polynomial direction algorithm then follows and is believed to be an orignial application to this field [Ref. 10].



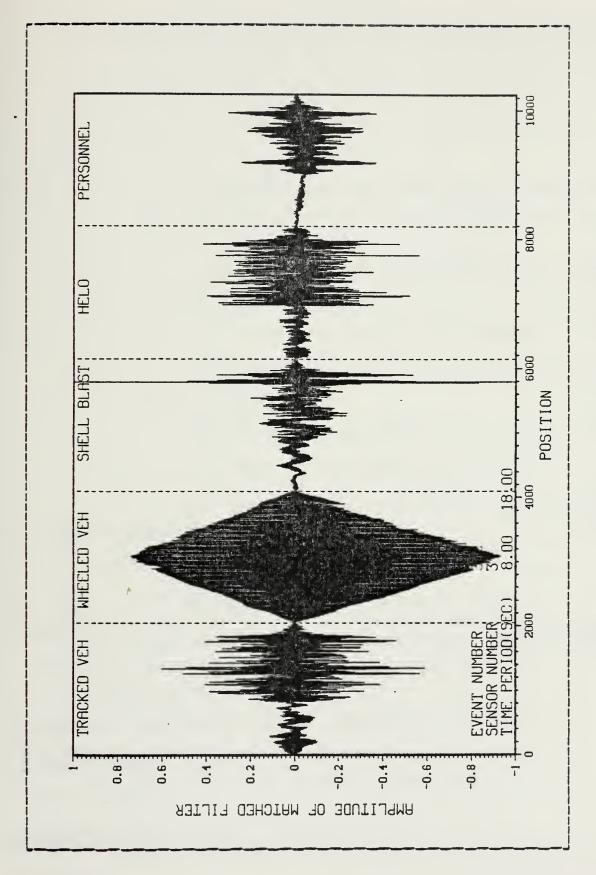


Figure 5.1 Two Target Matched Filter Response



- 1. Multiple Target Direction Phase Difference Algorithm
 - $T_{i,j}$ the arrival time of the seismic signal at the I-th sensor for the J-th target class
 - θ_i the angle of the I-th sensor to the x-axis
 - $D_{\mbox{\scriptsize ji}}$ the distance from the origin to where the wave front of the J-th target passes the I-th sensor
 - ${\tt Tc_{j}}$ the time when the J-th target class passes the orgin
 - X_i , Y_i the position of the I-th sensor
 - $\mbox{\bf B}_{\mbox{\scriptsize j}}$ the arrival angle of the seismic wave for the J-th target class
 - V the seismic wave velocity
 - R the radius of the sensor ring
 - I the sensor number where I has integer values form one to nine
 - J the J-th target class
 - N the number of sensors in the ring

Figure 5.2 illustrates these parameters and their interdependence. The derivation of the algorithm follows:

$$\theta_i = 2 (I - 1) /N$$

where zero degrees is set parrallel to the x - axis

 $X_i = R\cos\theta_i$

 $Y_i = Rsin \theta_i$

 $D_{ii} = Rcos(\theta_i - B_{i})$



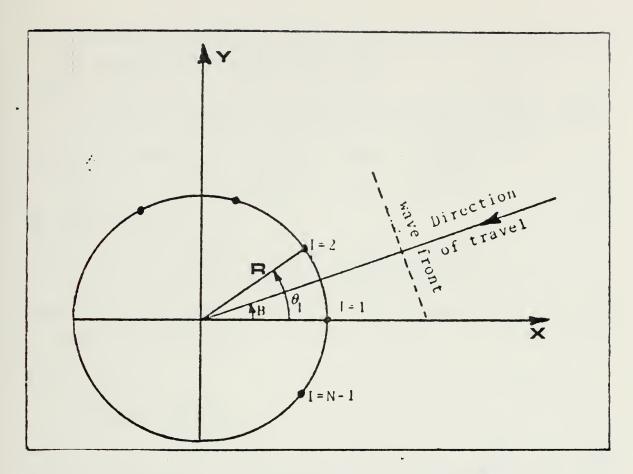


Figure 5.2 Circular Sensor Array Geometry

or equivalently
$$D_{ji} = (X_i) \cos B_j + (Y_i) \cos B_j$$

$$Tc_j = (1/N) \sum_{i=1}^{N} T_{ji}$$

$$\tau_{ji} = Tc_j - T_{ji}$$

$$V = D_{ji} / \tau_{ji}$$

$$\mathbf{V} = ((\mathbf{X}_{i}) \cos \mathbf{B}_{j} + (\mathbf{Y}_{i}) \sin \mathbf{B}_{j}) / (\mathbf{T}\mathbf{c}_{j} - \mathbf{F}_{ji})$$
 (5.1)

Since the wave velocity can be assumed to be constant when passing all sensors, then for any sensor I and K where I \neq K, equation 5.1 leads to



$$((X_i)\cos B_j + (Y_i)\sin B_j)/(Tc_j - T_{ji}) = ((X_i)\cos B_j + (Y_i)\sin B_j)/(Tc_j - T_{jk})$$

Where I # K

Now solving for the arrival angle $B_{\rm jik}$ of the wave

or equivalently;

$$B_{jik} = \arctan(((Tc_j - T_{jk}) \cos \theta_i - (Tc_j - T_{ji}) \cos \theta_k) / (Tc_j - T_{ji}) \sin \theta_k - (Tc_j - T_{jk}) \sin \theta_i))$$

Where the values of $T_{j\,i}$ and $T_{j\,k}$ are returned values from the matched filter routine.

Now:

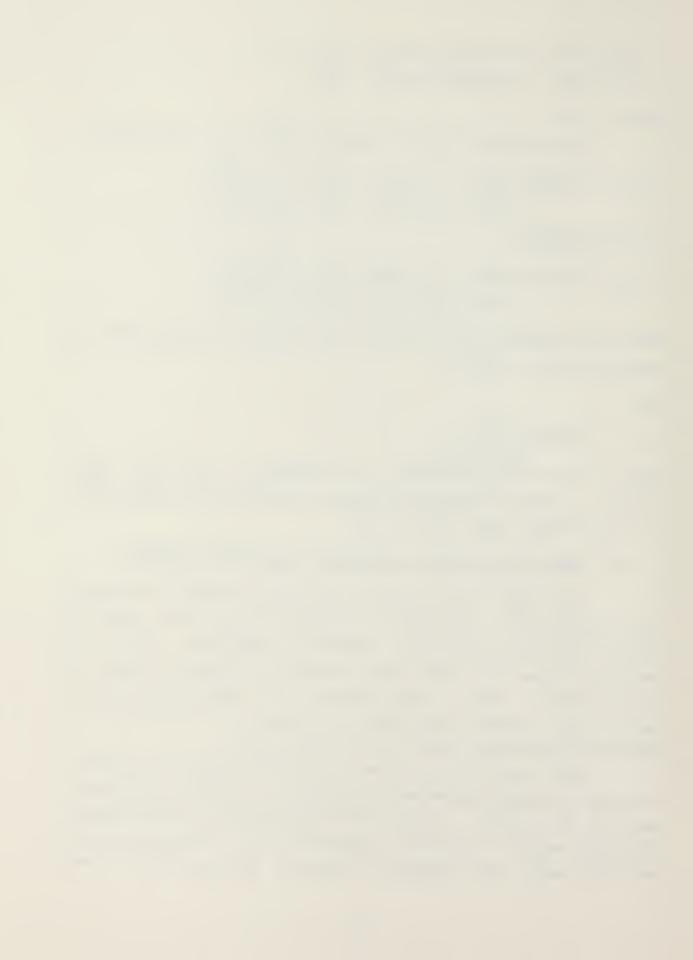
$$B_{j} = (1/(N)^{2}) \sum_{i=1}^{N} \sum_{k=1}^{N} B_{jik}$$

where B_{j} is the direction in radians to the J-th class target. For the multiple direction routine as implemented, it has values from one to five.

2. Least Mean Square Polynomial Direction Finding

The least mean square direction finding algorithm was developed in response to problems encountered with the phase difference direction finding algorithm. This new method is base on a least mean squares polynomial curve fit of the sensor data. This approach was selected since the least mean squares polynomial provides for best fit or a maximum likelihood curve fit for noisy data.

The least mean squares direction finding algorithm, as with the phase difference algorithm, assumes the seismic wave to be planar. Figure 5.2 illustrates the parameters for this model. Once the assumption of a plannar seismic wave is made, the expected relation between the arrival



angle, relative delay times and sensor positon in the circular array, can be made. Figure 5.3 illustrates these relations for a nine sensor circular array with a seismic wave arriving at zero degrees. Notice that the relative delay times have been scaled to be from zero to one.

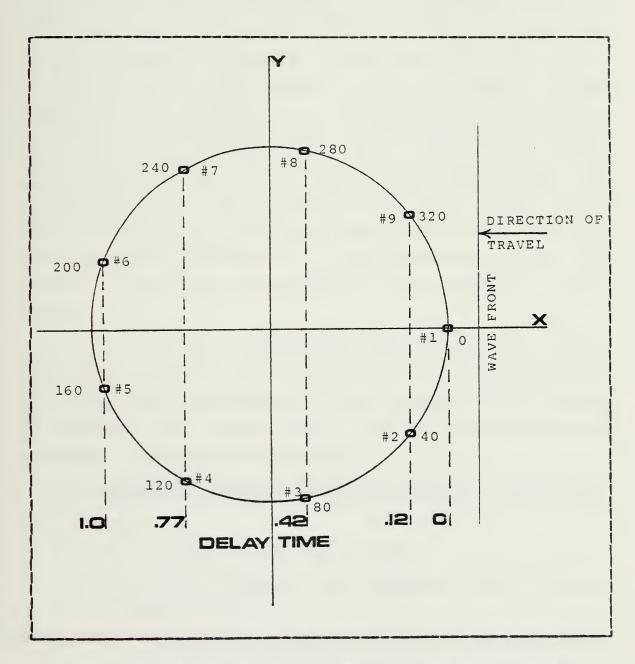


Figure 5.3 Relative Delay Times in a Nine Sensor Ring



Since the sensors in a nine sensor ring are at increments of forty degrees relative to the x-axis, a correlation can be seen between the relative time delays at each . sensor and the arrival angle of the seismic wave. A plot can now be made to illustrate the relationship of sensor angles versus relative time delays. Figure 5.3 is a plot for a nine sensor ring with a plannar seismic wave arriving at zero degrees. Figure 5.4 shows that for a wave arrival angle of 160 degrees, the delay time at sensor number five will be zero. It can be seen that the fitting of these ideal data points with a least mean squares polynomial will produce an equation for a curve whose minimum value is also at the arrival angle of the seismic wave. The minimum degree of the polynomial to fit this ideal data is four. This results from noting that the curve in figure 5.4 has three curve inflections. For experimental data, this minimum curve point corresponds to the predicted arrival angle.

Polynomials of degrees higher than four may be expected to enhance arrival angle errors since the polynomial would distort to fit noisy data. Least means squares polynomials of degree two and three however, may be useful in reducing curve sensitivity to one or two malfunctioning sensors or excessively noisy data.

3. <u>Least Mean Squares Polynomial Algorithm Derivation</u> Let:

N - number of data point pairs

 \mathbf{Y}_{i} - the observed or experimatal data position values

 \mathbf{x}_i - independent degree values with a range of 0 to 360 degrees



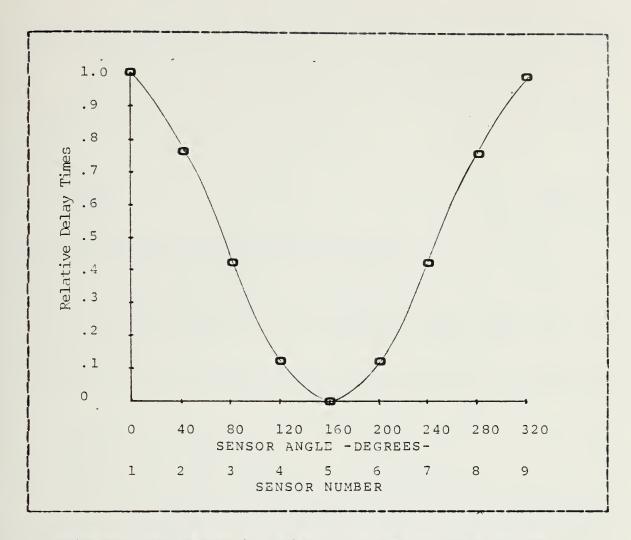


Figure 5.4 Relative Time Delay versus Sensor Angle

 P_i - dependent predicted delay time values found by the least mean squares polynomial

 \mathbf{A}_i , \mathbf{B}_i - coefficient values for the system of simultaneous equations

 ${\bf E}_{\hat{\bf i}}^{-}$ error between the experimental data values and predicted values of time delays

S - sum of the square errors between each data and predicted points

The derivation of the least mean squares polynomial follows: $S = E_1^2 + E_2^2 + E_3^2 + \dots + E_N^2$



$$P_{i} = A_{0} + A_{1}X_{i} + A_{2}X_{i}^{2} + A_{3}X_{i}^{3} + \dots + A_{n}X_{i}^{n}$$
 (5.2)

$$E_i = Y_i - P_i \tag{5.3}$$

$$S = \sum_{i=1}^{N} E_i^2$$
 (5.4)

Combining equations 5.2 and 5.3 yields

$$E_i = Y_i - A_0 - A_1 X_i - A_2 X_1^2 - \dots - A_n X_i^n$$

where n is the degree of the polynomial such that N > n + 1 and 1 < i < N

Equation 5.4 now becomes, after substituting for E, equation 5.5

$$S = \sum_{i=1}^{N} (Y_i - A_0 - A_1 X_i - A_2 X_i^2 - \dots - A_n X_i^n)^2$$
 (5.5)

To find the minimum of the sum of the squares expressed by equation 5.5, the partial derivatives of S with respect to all of the coefficients are taken. At the minimum, these partial derivatives all vanish.

$$\frac{\partial S}{\partial A_0} = 0 = \sum_{\substack{i=1 \\ N}}^{N} 2 (Y_i - A_0 - A_1 X_i - \dots - A_n X_i^n) (-1)$$

$$\frac{\partial S}{\partial A_1} = 0 = \sum_{\substack{i=1 \\ N}}^{N} 2 (Y_i - A_0 - A_1 X_i - \dots - A_n X_i^n) (-X_i)$$

$$\frac{\partial S}{\partial A_n} = 0 = \sum_{\substack{i=1 \\ N}}^{N} 2 (Y_i - A_0 - A_1 X_i - \dots - A_n X_i^n) (-X_i^n)$$

Dividing by two and rearranging gives n + 1 normal simultaneous equations. Expressed in matrix notation these equations become



$$\begin{bmatrix} \mathbf{N} & \sum \mathbf{X}_{\mathbf{i}} & \sum \mathbf{X}_{\mathbf{i}}^{\mathbf{2}} & \dots & \sum \mathbf{X}_{\mathbf{i}} \\ \sum \mathbf{X}_{\mathbf{i}} & \sum \mathbf{X}_{\mathbf{i}}^{\mathbf{2}} & \sum \mathbf{X}_{\mathbf{i}}^{\mathbf{3}} & \dots & \sum \mathbf{X}_{\mathbf{i}} \\ \sum \mathbf{X}_{\mathbf{i}}^{\mathbf{2}} & \sum \mathbf{X}_{\mathbf{i}}^{\mathbf{3}} & \sum \mathbf{X}_{\mathbf{i}}^{\mathbf{4}} & \dots & \sum \mathbf{X}_{\mathbf{i}} \\ & \vdots & & \vdots & & \vdots \\ \sum \mathbf{X}_{\mathbf{i}}^{\mathbf{n}} & \sum \mathbf{X}_{\mathbf{i}+1}^{\mathbf{n}} & \sum \mathbf{X}_{\mathbf{i}+2}^{\mathbf{n}+2} & \dots & \sum \mathbf{X}_{\mathbf{i}}^{\mathbf{n}} \end{bmatrix} \begin{bmatrix} \mathbf{A}_{\mathbf{0}} \\ \mathbf{A}_{\mathbf{1}} \\ \mathbf{A}_{\mathbf{2}} \end{bmatrix} = \begin{bmatrix} \sum \mathbf{Y}_{\mathbf{i}} \\ \sum \mathbf{X}_{\mathbf{i}}^{\mathbf{i}} \mathbf{Y}_{\mathbf{i}} \\ \sum \mathbf{X}_{\mathbf{i}}^{\mathbf{2}} \mathbf{Y}_{\mathbf{i}} \\ \sum \mathbf{X}_{\mathbf{i}}^{\mathbf{2}} \mathbf{Y}_{\mathbf{i}} \\ \sum \mathbf{X}_{\mathbf{i}}^{\mathbf{n}} \mathbf{Y}_{\mathbf{i}} \end{bmatrix}$$

The coefficient matrix in the above system of equations can be solved for. The minimum value for P can then be found. The X corresponding to this minimum value is declared to be the predicted arrival angle of the seismic wave. [Ref. 10]

4. Adaptive Target Direction Finding

An adaptive method is used to improve the directions found for all single target cases. To enhance the resolution accuracy of the peak filter output position, the single target seismic data is copied into the filter data section for its class of target. Subsequently, when the matched filter routine is called for each sensor, filtering is performed with sensor data from that time period.

5. <u>Software features of the Multiple Target Direction</u> Routine

As first addressed in the previous chapter, the multiple direction routine allows for the selection of the number of data points to be used from the 1024 size buffer of sensor data. This was included to investigate the algorithm's performance for various data/filter window sizes and to allow an option for reduced CPU time utilization for interactive program runs. Tabular output from this routine includes the event number, the time period for the data, identification and directions for up to five target classes, and up to four simulated target specifications. Also, notice that the routine adapts to the number of sensors



specified for the ring. This was necessary since the experimental data included sensor rings of six and nine vertical sensors.

C. MULTIPLE TARGETS OF THE SAME TARGET CLASS

A limitation on the system, as presented up to this point, is its inablity to engage multiple targets of the same class. Discrete targets may appear as separate entities since the seismic signals generated by the two same class targets are not likely to be incident at the same time. This situation is greatly complicated for continuous time signals, such as tracked or wheeled vehicles. For such continuous time targets, mutual distortion would be the likely result.

An algorithm is needed to identify these multiple peaks without erroneously declaring elements in the same peak as targets. By using only the positive half of the matched filter output and performing data smoothing on the remaining points, a curve with its number of peaks equalling the number of targets present could be generated. Numerical methods for curve fitting or interpolation are available [Ref. 11.]. Polynomial curve fitting, Sterling's method and variations of Newton's method are only a few of the possible approaches applicable for equally spaced data. By differentiating the smoothed data, the peaks and valleys of the filter output can be found. The height of the curve corresponding to the points where the derivative is zero can now be compared to the selected matched filter threshold. excludes the valley points and leaves the points remaining which correspond to the relative times of the same class targets.

This method is constrained by resolution of close proximity, time limited targets and near phase synchronization



of continuous targets. The variables to be optimized through experimentation may be the degree of the smoothing of the curve data and the exclusion of erroneous valleys associated with the same target's data.



VI. ANALYSIS OF SEISMIC DATA

Analysis of the simulated and experimental seismic data will be conducted as detailed in Table II. Table III lists the matched filter contents for the simulated data. Table IV is the test plan for the experimental data. Table V lists the matched filter signals used for the experimental data analysis.

Table VI summarizes the results of the simulted and experimental data runs for direction finding. The window size, used for all multiple target direction finding results, was 300. Table VII lists the events in which targets were missed or incorrectly identified.

The time domain phase difference directions found, are not presented for the reasons noted earlier. Errors of up to eighty degrees were not uncommon with this method.

Each event run will be accompanied by the following graphic output:

- 1. Least Mean Square Initial Direction
- Matched Filter Response
- 3. Amplitude Response
- 4. Amplitude Response of any Malfunctioning Sensor
- 5. Frequency Response
- 6. Least Mean Squares Polynomial Curve Fitting (LMSP)
 Using Matched Filter Outputs
- 7. Multiple Target Direction Summary Resulting from Least Mean Squares Curve Fitting



TABLE II
Test Plan for Simulated Data

Event	#Sen	#Igts	Frequency	<u>Amplitude</u>	Direction
001	9	1	10	3000	0
001	9	1	10	3000	40
001	9	1	10	3000	120
001	9	1	10	3000	240

TABLE III
Matched Filter for Simulated Targets

Filter	Frequency	<u>Amplitude</u>	Direction
Tracked V	<i>l</i> eh 30	2000	0
Wheeled N	/eh 10	2000	0
Shell bla	ast data from	event #383	0
Helicopte	er 15	2000	0
Personnel	L 20	2000	0



TABLE IV
Test Plan for Experimental Data

Event	#Sen	#Tqts	Dir	Target	Distance
383	9	1	0	Shot	5KM
382	9	1	0	Shot	5Km
372	6	1	315	Helicopter	5 - 15KM
375	6	1	0	Tank	5 - 0Km
374	6	1	315	Helicopter	15 - 5KM
302	6	1	0	Mortar	1 KM
314	6	1	315	LVT	4 - 5KM
354	б	5	0 225 315 0	105mm How 175mm Gun LVT M-60 Pank	5 Km 4 KM 4 - 5 KM 4 - 5 KM
	383 382 372 375 374 302 314	383 9 382 9 372 6 375 6 374 6 302 6 314 6	383 9 1 382 9 1 372 6 1 375 6 1 374 6 1 302 6 1 314 6 1	383 9 1 0 382 9 1 0 372 6 1 315 375 6 1 0 374 6 1 315 302 6 1 0 314 6 1 315 354 6 5 0 225	383 9 1 0 Shot 382 9 1 0 Shot 372 6 1 315 Helicopter 375 6 1 0 Tank 374 6 1 315 Helicopter 302 6 1 0 Mortar 314 6 1 315 LVT 354 6 5 0 105mm How 225 175mm Gun LVT

TABLE V
Matched Filter for Experimental Data

t <u>Used</u> as Filter
e (backround noise)
e (backround noise)



TABLE VI Summary of Direction Finding Results

Event	#Sen	#Tqts	<u>Distance</u>	Dir İ	<u>Initia</u>	<u>l</u> Zerror	LMSP DIE	<u> </u>
001	9	1	N/A	0	N/A		0	0
001	9	1	N/A	40	N/A		40	0
001	9	1	N/A	120	N/A		120	0
001	9	1	N/A	240	N/A		240	0
383	9	1	5KM	0	4 (3)	1.1	28 (4	7.8
382	9	1	5KM	0 -	-14 (4)	3.9	- 6	1.67
372	6	1	5 - 15 KM	315	-32	3.6	-59	3.9
375	6	1	5 - OKM	0	312	13.3	FAII	.ED
374	6	1	15 - 5 KM	315	- 59	4.0	FAII	.E D
302	6	1	1KM	0	4	1.1	6	1.7
354	6	5	5KM 4 KM 4 - 5KM 4 - 5KM	0 225 315 0	-3 291 FAIL FAIL		3 FAII	.ED.8

*Note: The matched filter threshold was set at .9 for all single targets and .6 for all multiple target events. Brackets indicate the use of other than a second degree polynomial.

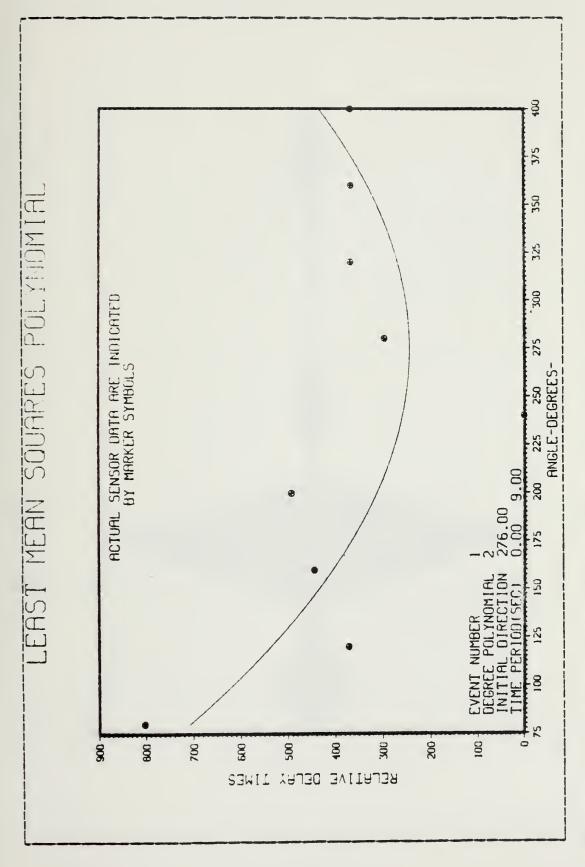


TABLE VII
Missed or Incorrectly Identified Targets

<u>Event</u>	#Targets	Target	Nature of Errors
375	1	Tank	 Matched filter was not based on a high S/N sample signal
374	1	Helo	2. Small seismic signal amplitudes
354	1	LVT Tank	3. Malfunctioning sensor(s)
	1 175 m i Gun		Sever clipping distortion of input signal

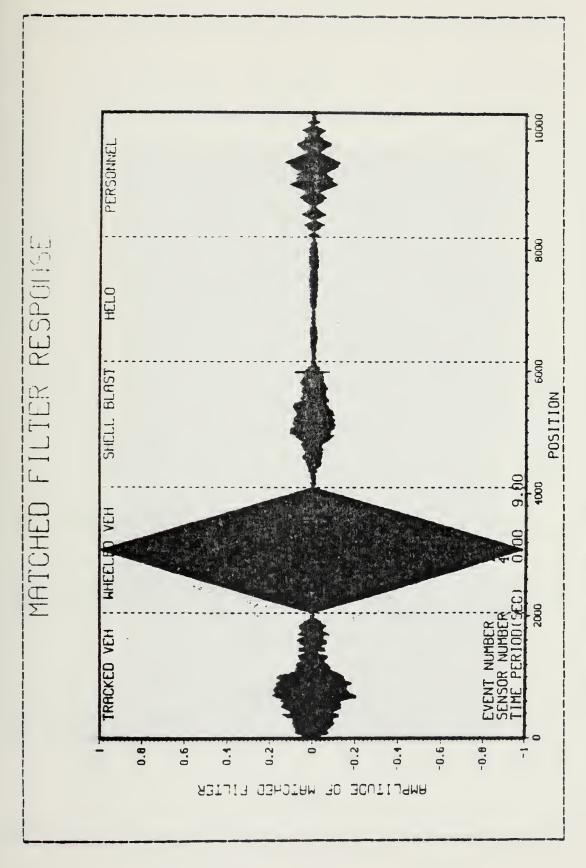
Note: The numbered error sources apply to all events listed.





Sample Least Mean Squares Initial Direction for Event 001 Figure 6.1



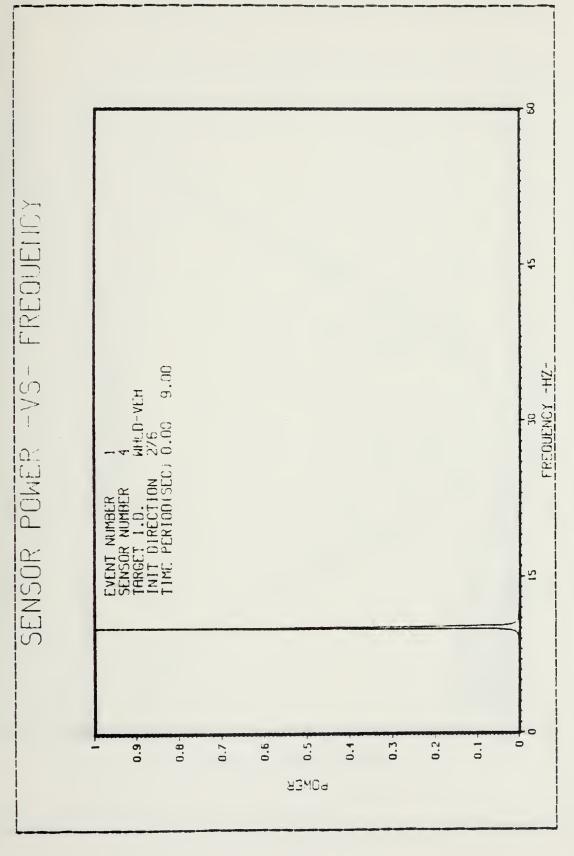


Sample Matched Pilter Response for Event 001 Figure 6.2



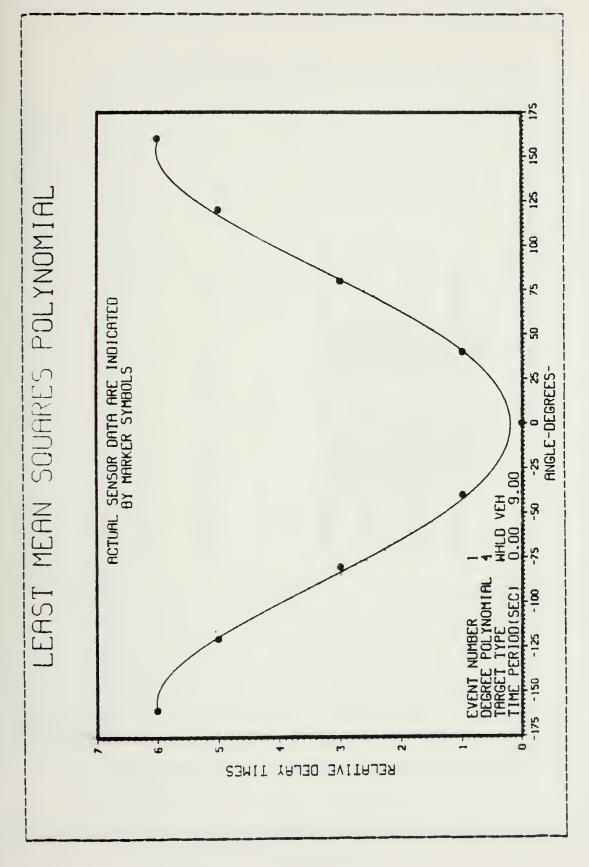
Sample Amplitude Response for Event 001 Figure 6.3





Sample Frequency Response for Event 001 Figure 6.4





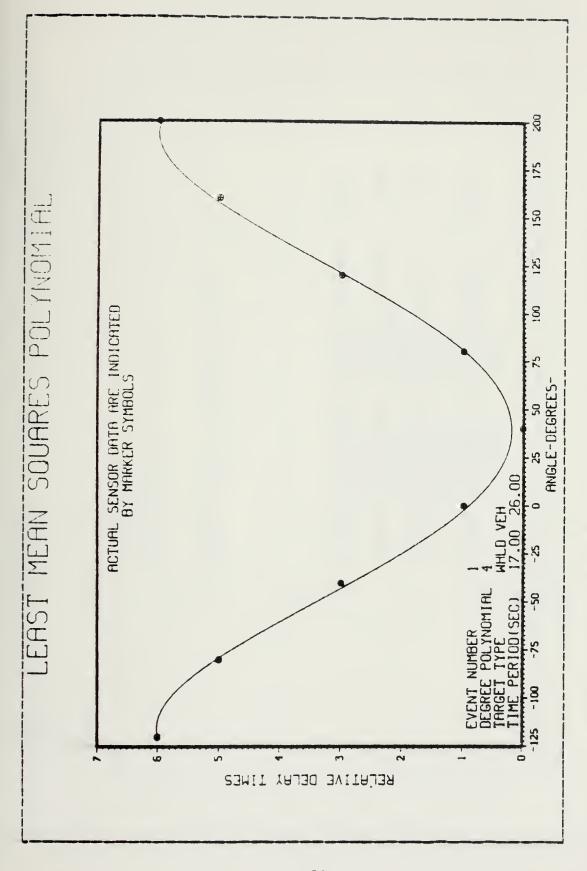
LMSP Matched Filter Direction for Event 001 Pigure 6.5



THRGET - MATCHED FILTER OUTPUT EVENT NUMBER 1	TIME PERIOD(SEC) 0.00 9.00 WHEELED VEHICLE DIRECTION - 0.00	SIMULATED TRKD VEHICLE TARGET FREQUENCY 0.00 RMPLITUDE 0.0000 DIRECTION 0.0000		띹	SIMULATED PERSONNEL TARGET FREQUENCY 0.00 RMPLITUDE 0.0000 DIRECTION 0.0000	
IMBER	0	SIMULATED TRKD VEHICL AMPLITUDE DIRECTION	LOCAL TODE	띹	.급	

LMSP Multiple Target Direction Summary for Event 001 Figure 6.6





LNSP Matched Filter Direction for Event 001 Figure 6.7

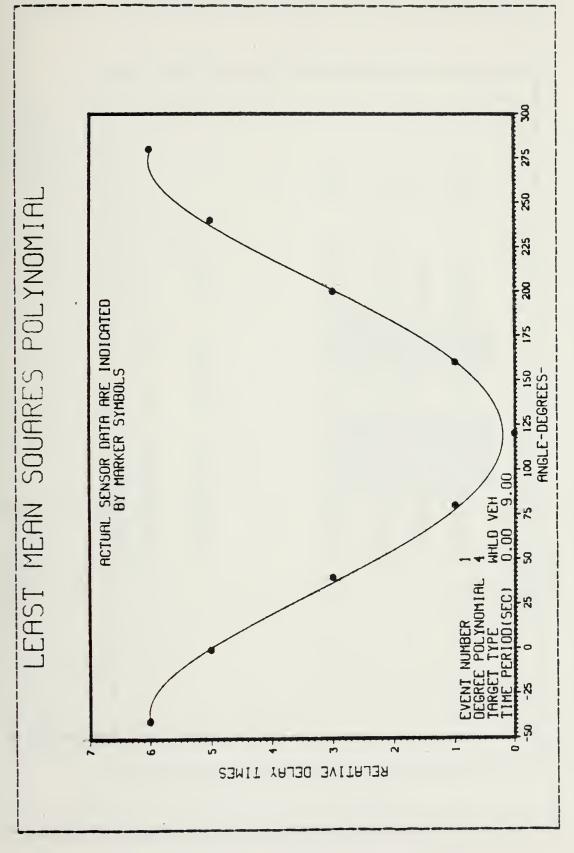


MATCHED FILTER OUTPUT MULTIPLE TARGET

	4CY 0.00	4CY 10.00	0.00
EVENT NUMBER 1 TIME PERIODISECI 17.00 26.00 WHEELED VEHICLE DIRECTION - 40.00	SIMULATED TRKD VEHICLE TARGET FREQUENCY	SIMULATED VEHICLE TARGET FREQUENCY A0.0000 BIRECTION TO THE TARGET FREQUENCY A0.0000 BIRECTION TO TARGET FREQUENCY SIMULATED HELICOPTER TARGET FREQUENCY	HMPLITUDE 0.0000 DIRECTION 40.0000 SIMULATED PERSONNEL TARGET FREQUENCY AMPLITUDE 0.0000 DIRECTION 0.0000

LMSP Multiple Target Direction Summary for Event 001 Figure 6.8





LMSP Matched Filter Direction for Event 001 Pigure 6.9

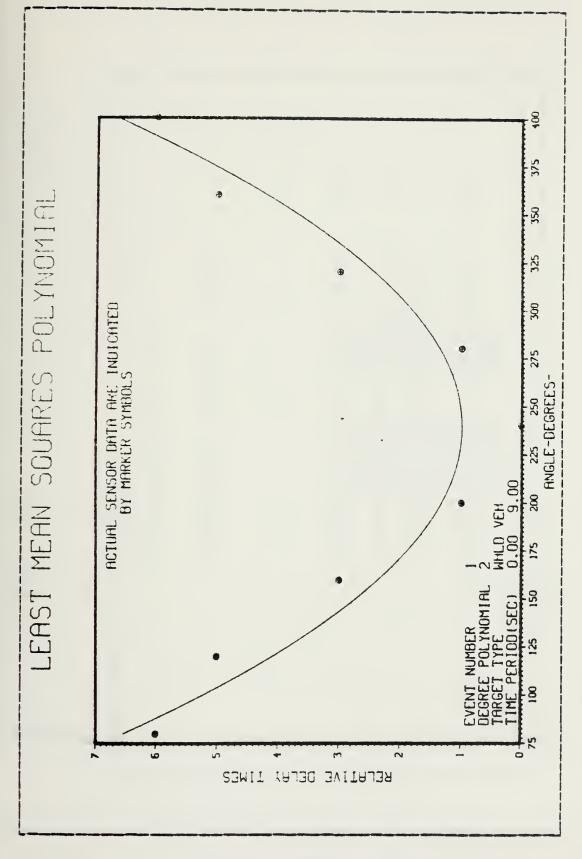


- MATCHED FILTER OUTPUT MULTIPLE TARGET

		10.00	0.00	0.00	0.00
EVENT NUMBER 1 TIME PERIOD(SEC) 0.00 9.00	WHEELED VEHICLE DIRECTION - 120.00	SIMULATED TRKD VEHICLE TARGET FREGUENCY RMPLITUDE 3000.0000	SIMULATED WHLD VEHICLE TARGET FREDUENCY AMPLITUDE 0.0000	SIMULATED HELICOPTER TARGET FREQUENCY AMPLITUDE 0.0000	SIMULATED PERSONNEL TARGET FREQUENCY AMPLITUBE 0.0000 DIRECTION 0.0000

LMSP Multiple Target Direction Summary for Event 001 Pigure 6.10





LMSP Matched Filter Direction for Event 001 Pigure 6.11

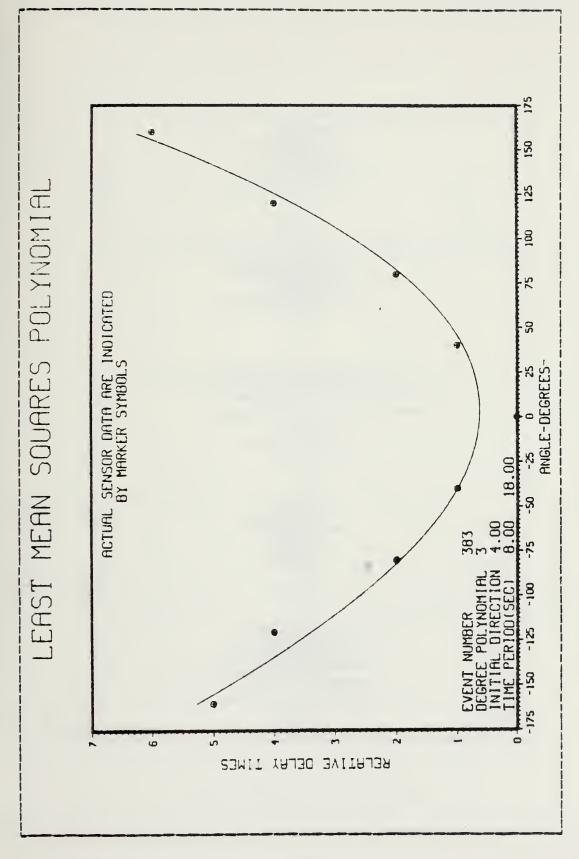


MULTIPLE TARGET - MATCHED FILTER OUTPUT

	0.00
EVENT NUMBER 1 TIME PERIODISECI 0.00 9.00 WHEELED VEHICLE DIRECTION - 240.00	SIMULATED TRKD VEHICLE TARGET FREQUENCY OINECTION SIMULATED WHLD VEHICLE TARGET FREQUENCY AMPLITUDE 3000.0000 SIMULATED HELICOPTER TARGET FREQUENCY AMPLITUDE 0.0000 SIMULATED PERSONNEL TARGET FREQUENCY AMPLITUDE 0.0000 DIRECTION 0.0000 SIMULATED PERSONNEL TARGET FREQUENCY AMPLITUDE 0.0000 DIRECTION 0.0000

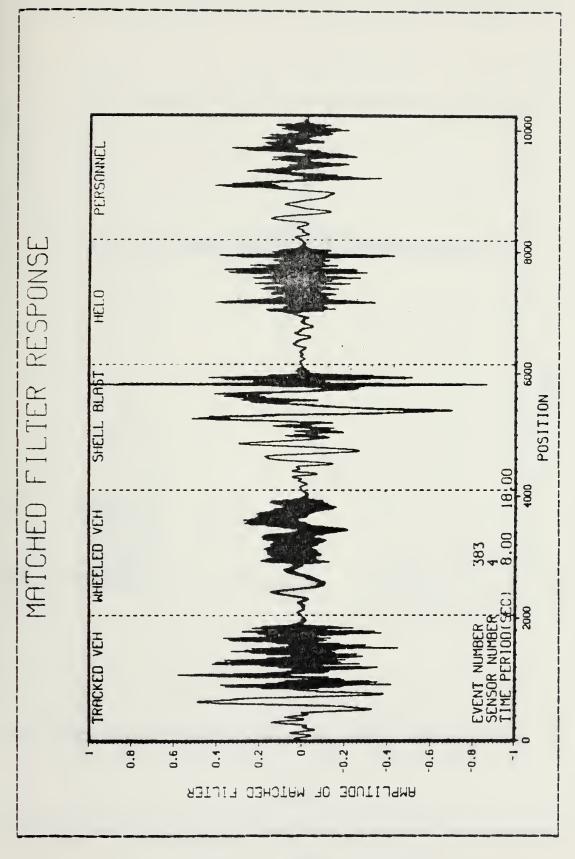
LMSP Multiple Target Direction Summary for Event 001 Figure 6.12





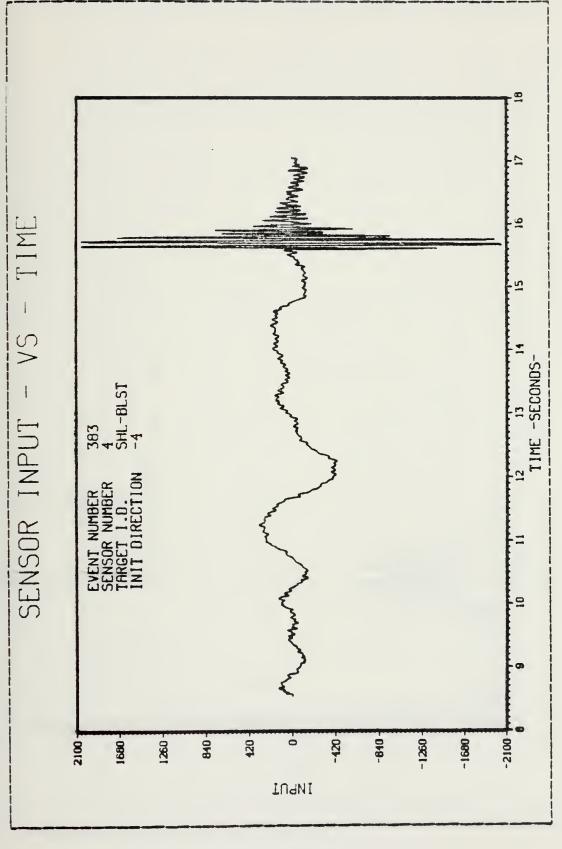
LMSP Initial Direction for Event 383 Figure 6.13





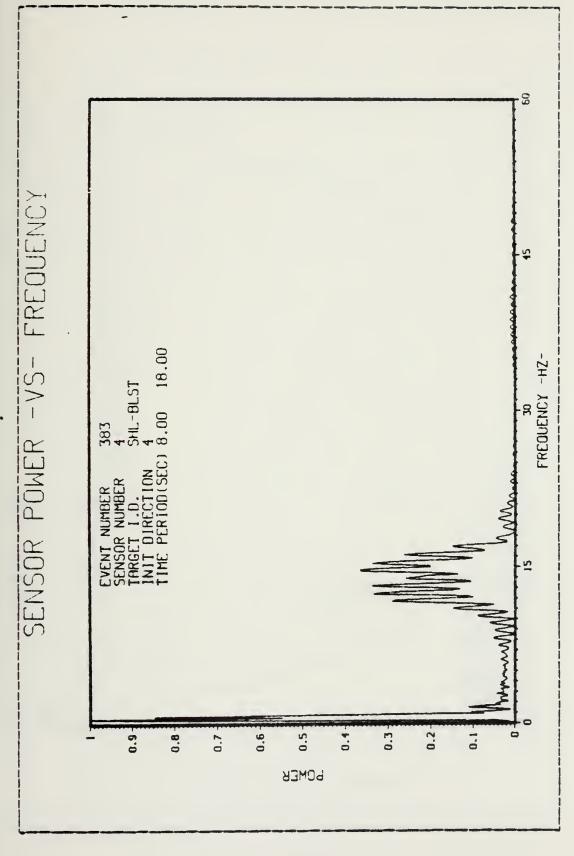
Matched Filter Response for Event 383 Pigure 6.14





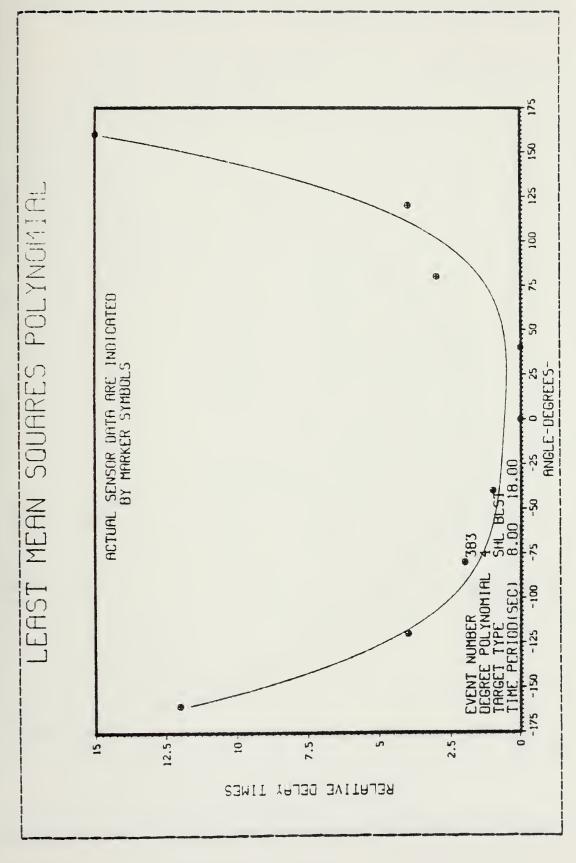
Pigure 6.15 Amplitude Response for Event 383





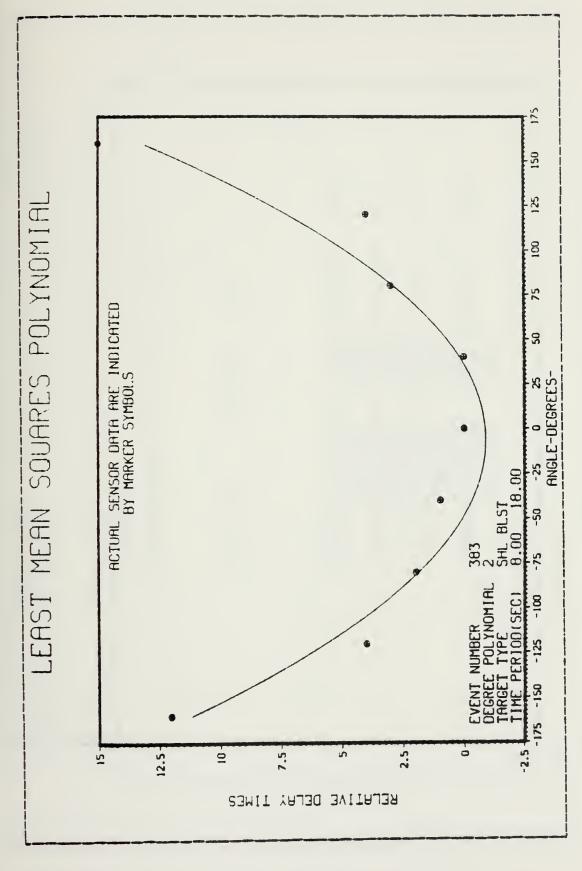
Pigure 6.16 Frequency Response for Event 383





Pourth Degree LMSP Matched Filter Direction for Event 383 Figure 6.17





Second Degree LMSP Matched Filter Direction for Event 383 Pigure 6.18

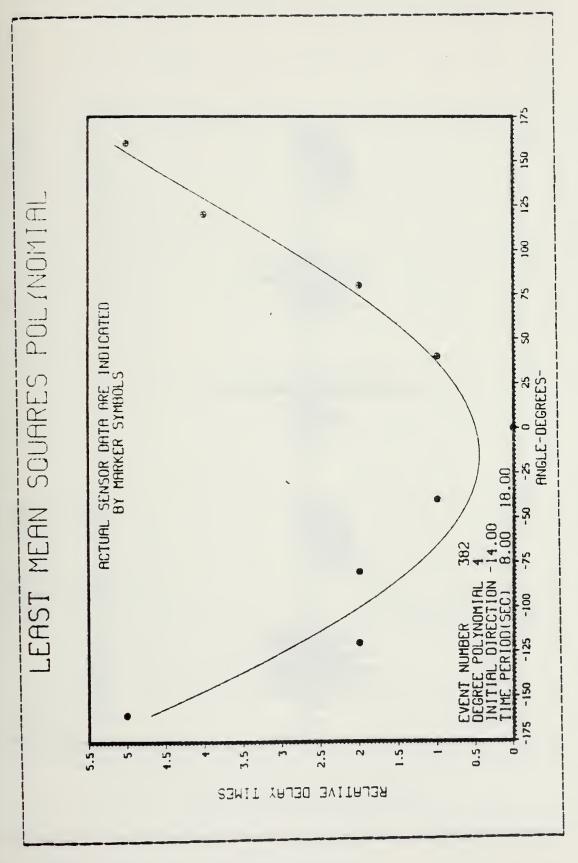


MULTIPLE TARGET - MATCHED FILTER OUTPUT

		00.0	0.00	0.00	0.00	
EVENT NUMBER 383 TIME PERIOD(SEC) 8.00 18.00	SHELL BLAST UIRECTION - 28.00	SIMULATED TRKD VEHICLE TARGET FREQUENCY AMPLITUDE 0.0000	SIMULATED WHLD VEHICLE TARGET FREQUENCY AMPLITUDE 0.0000	DIRECTION 0.0000 SIMULATED HELICOPTER TARGET FREGUENCY AMPLITUDE 0.0000	DIRECTION 0.0000 SIMULATED PERSONNEL TARGET FREGUENCY AMPLITUDE 0.0000 DIRECTION 0.0000	

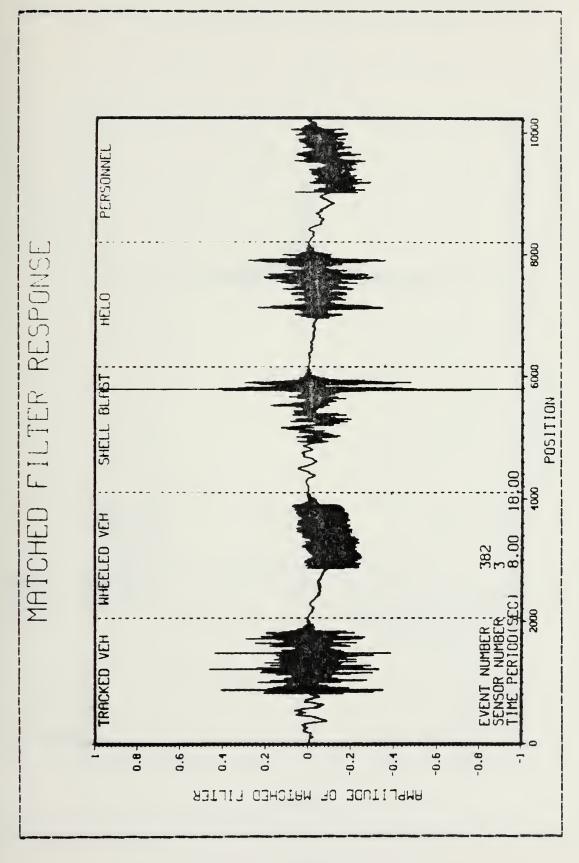
LMSP Multiple Target Direction Summary for Event 383 Figure 6.19





LMSP Initial Direction for Event 382 Figure 6.20





382 Natched Filter Response for Event Figure 6.21



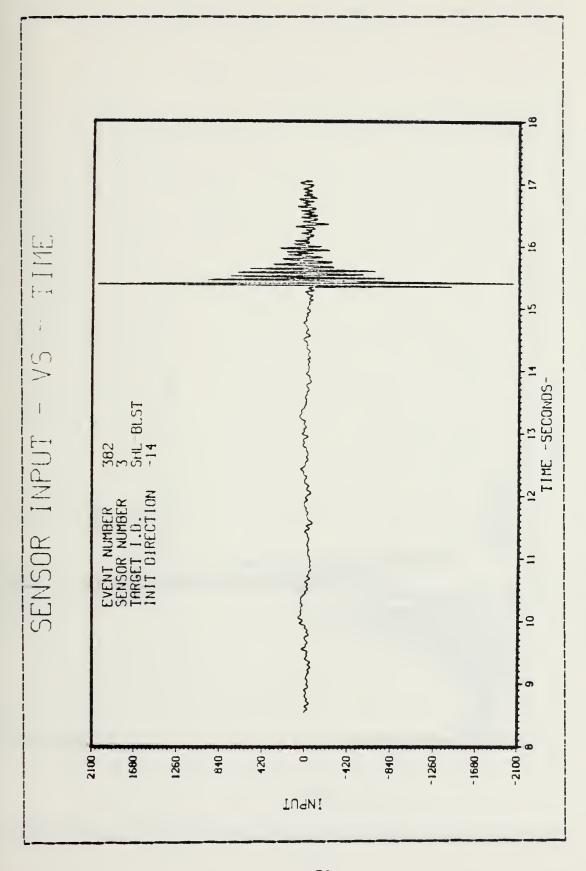


Figure 6.22 Amplitude Response for Event 382



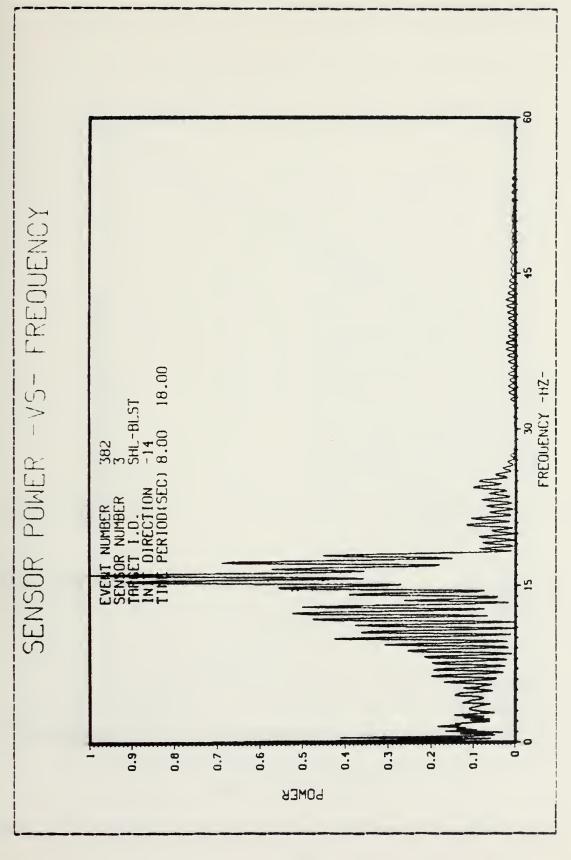
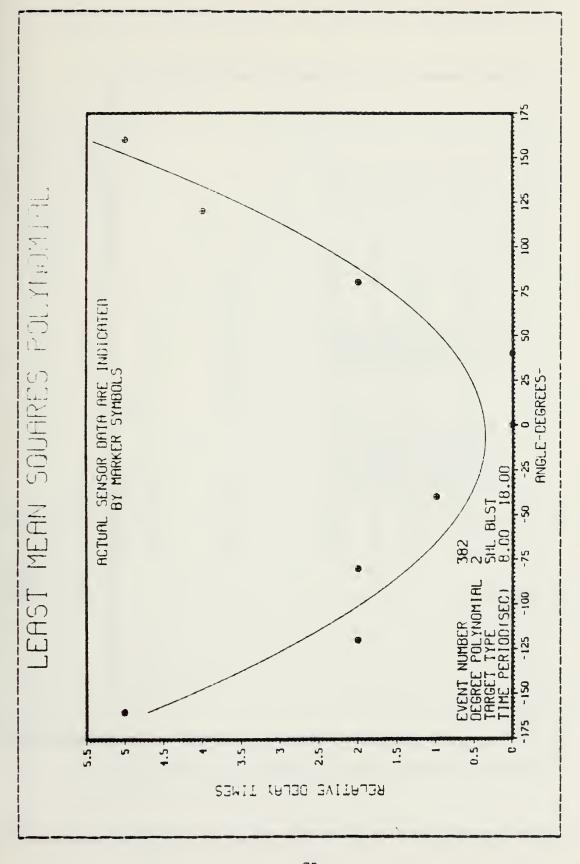


Figure 6.23 Frequency Response for Event 382





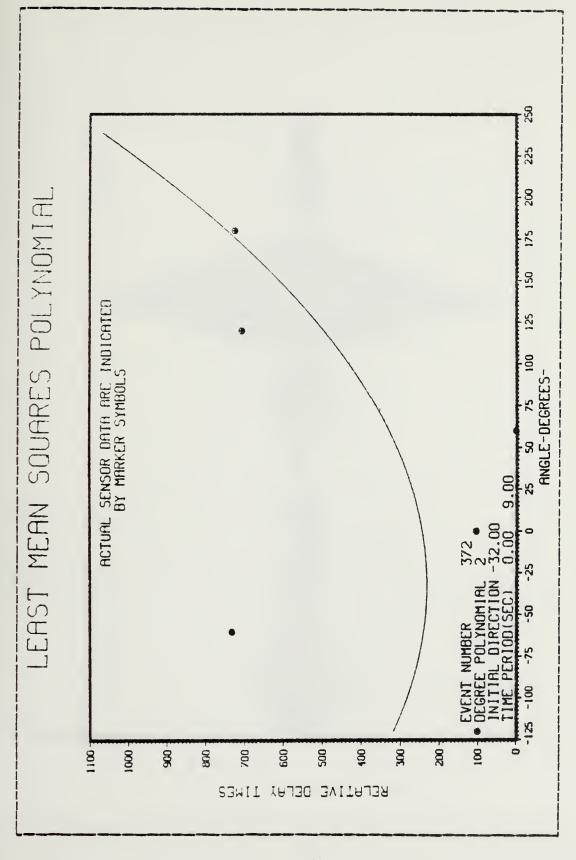
LNSP Matched Filter Direction for Event 382 Figure 6.24



MULTIPLE TAR	TARGET - 14F	TIME PERIOD(SEC) 8.00 18.00	SHELL BLAST DIRECTION6.00	SIMULATED TRKD VEHICLE TARGET FREQUENCY 0.00 AMPLITUDE 0.0000	DIRECTION 0.0000 SIMULATED WHICLE TARGET FREQUENCY 0.00 AMPLITUDE 0.0000	DIRECTION 0.0000 SIMULATED HELICOPTER TARGET FREQUENCY 0.00 AMPLITUDE 0.0000	BIRECTION 0.0000 SIMULATED PERSONNEL TARGET FREQUENCY 0.00 AMPLITUDE 0.0000 DIRECTION 0.0000
--------------	--------------	-----------------------------	---------------------------	--	--	--	--

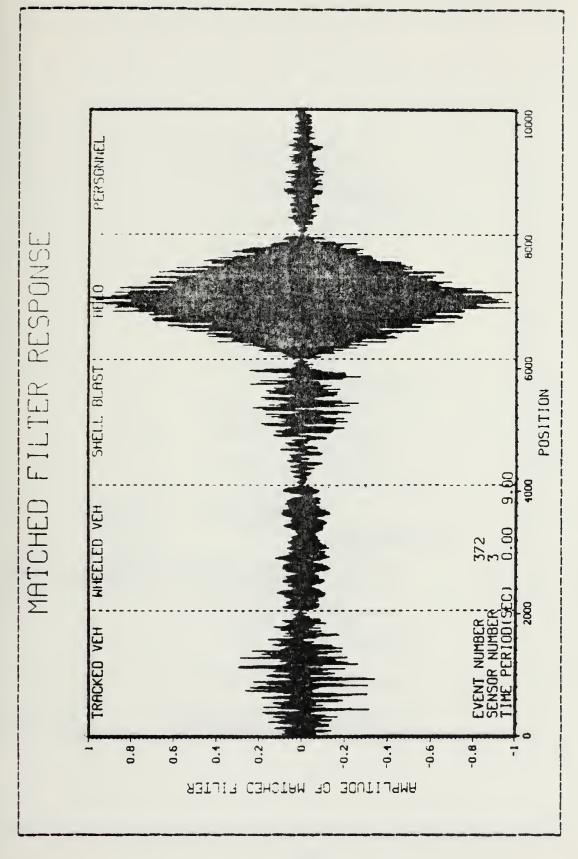
LMSP Multiple Target Direction Summary for Event 382 Pigure 6.25





LMSP Initial Direction for Event 372 Figure 6.26





Matched Filter Response for Event 372 Figure 6.27



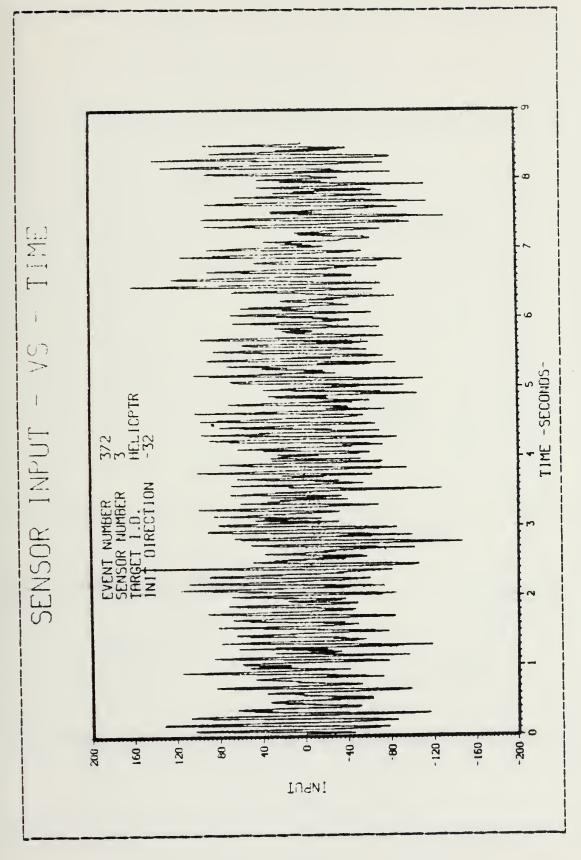
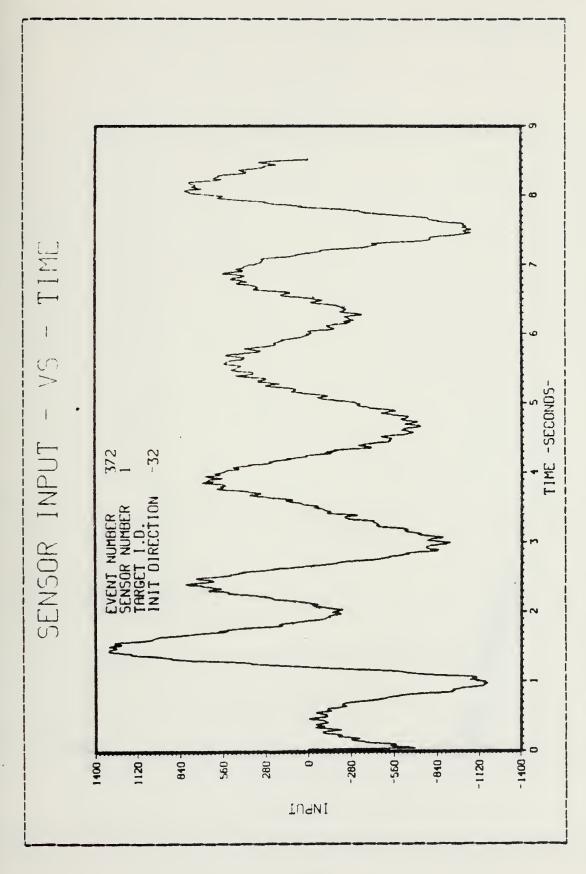


Figure 6.28 Amplitude Response for Event 372





Amplitude Response of Malfunctioning Sensor for Event 372 Pigure 6.29



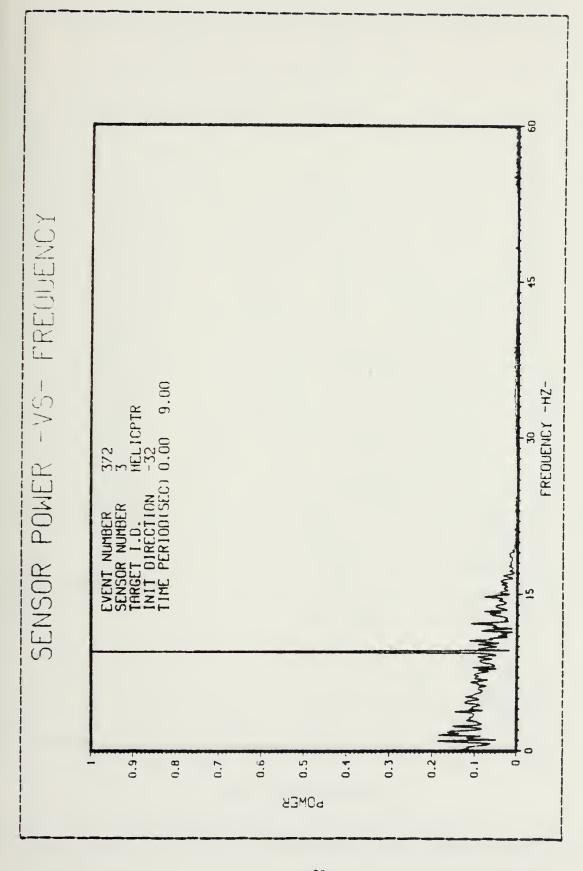
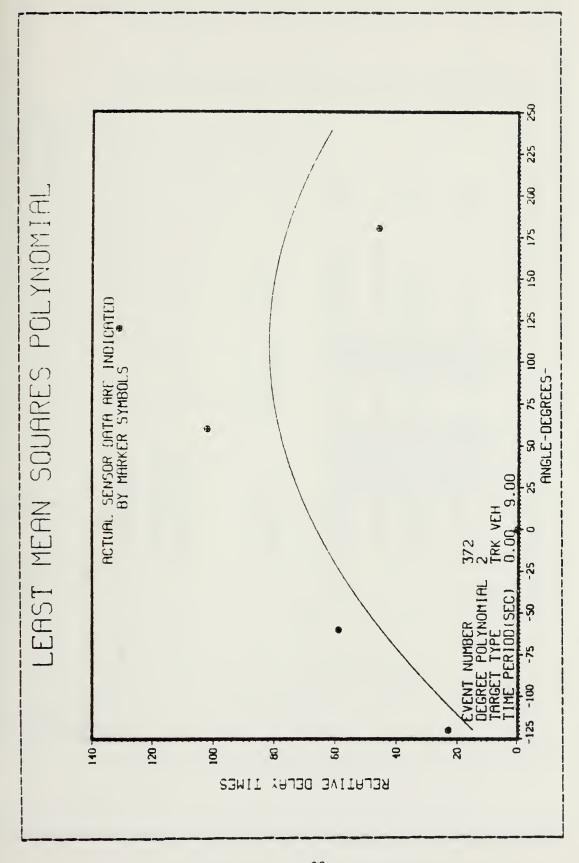


Figure 6.30 Frequency Response for Event 372





LMSP Matched Filter Direction for Event 372 Pigure 6.31

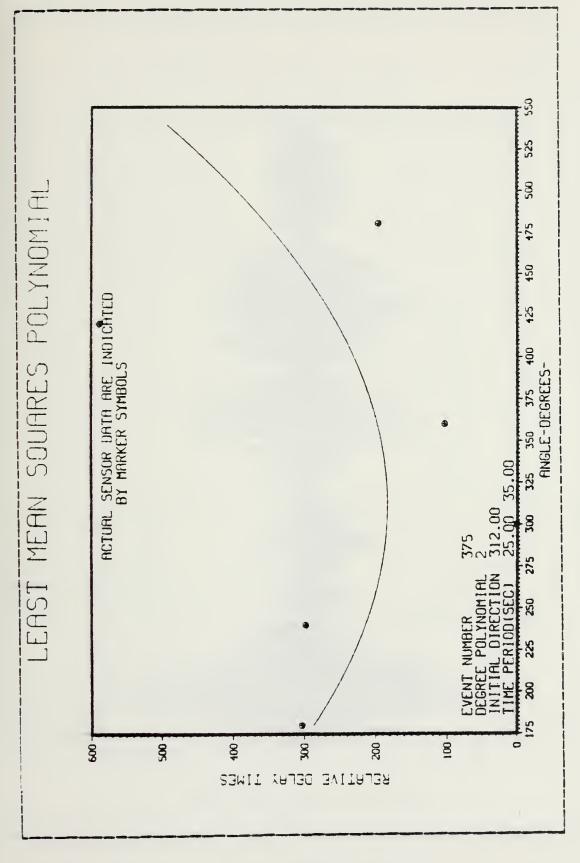


- MATCHED FILTER OUTPUT MULTIPLE TARGET

		0.00	0.00	0.00	0.00	
EVENT NUMBER 372 TIME PERIOD(SEC) 0.00 9.00 TRACKED VEHICLE DIRECTION - 0.00	HELICOPTER DIRECTION - 0.00	SIMULATED TRKD VEHICLE TARGET FREGUENCY AMPLITUDE 0.0000	SIMULATED MHLD VEHICLE TARGET FREDUENCY AMPLITUDE 0.0000	SIMULATED HELICOPTER TARGET FREGUENCY AMPLITUDE 0.0000	DIRECTION 0.0000 SIMULATED PERSONNEL TARGET FREQUENCY AMPLITUDE 0.0000 DIRECTION 0.0000	

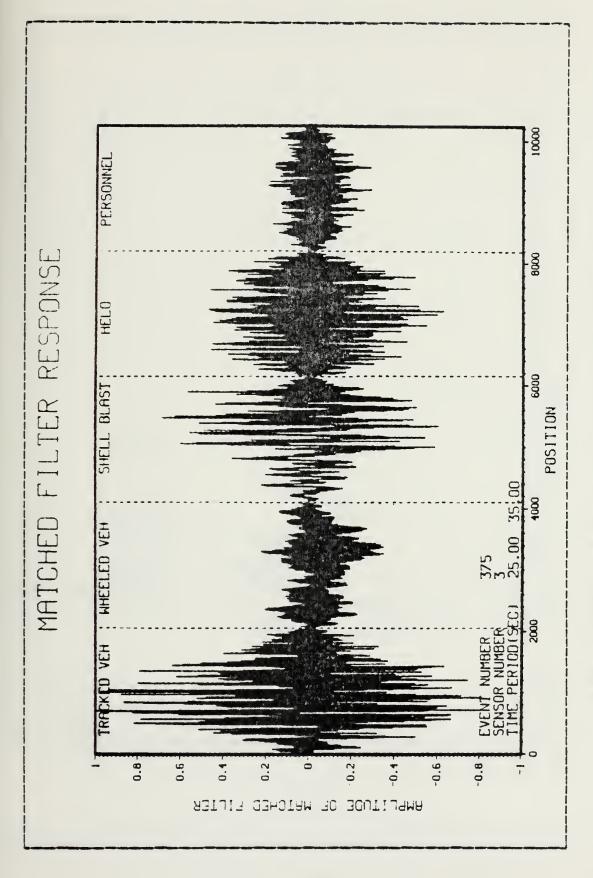
LMSP Multiple Target Direction Summary for Event 372 Figure 6.32





LMSP Initial Direction for Event 375 Pigure 6.33





Matched Filter Response for Event 375 Figure 6.34



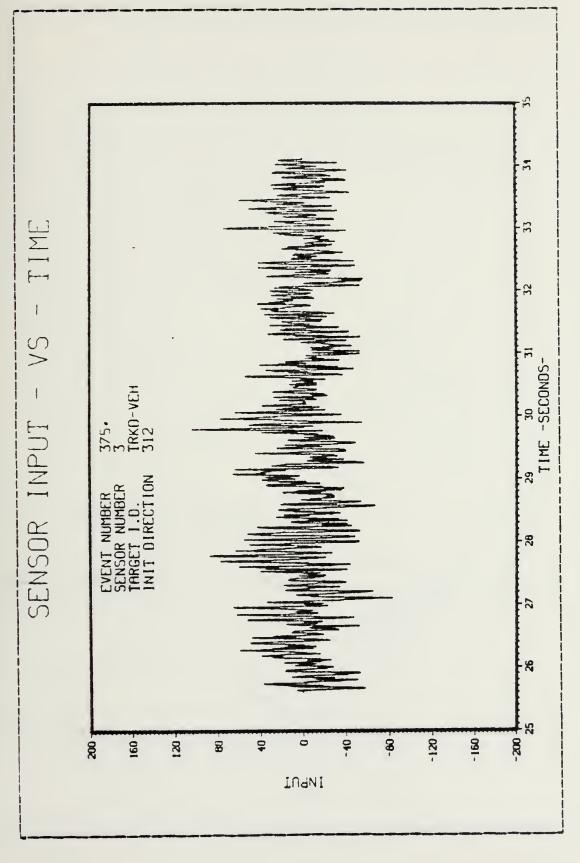
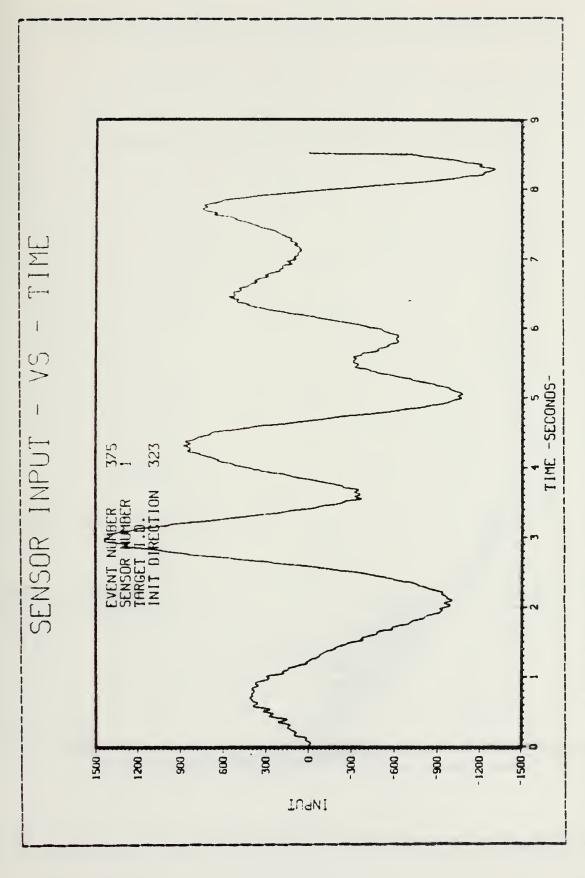


Figure 6.35 Amplitude Response for Event 375





Amplitude Response of Malfunctioning Sensor for Event 375 Figure 6.36



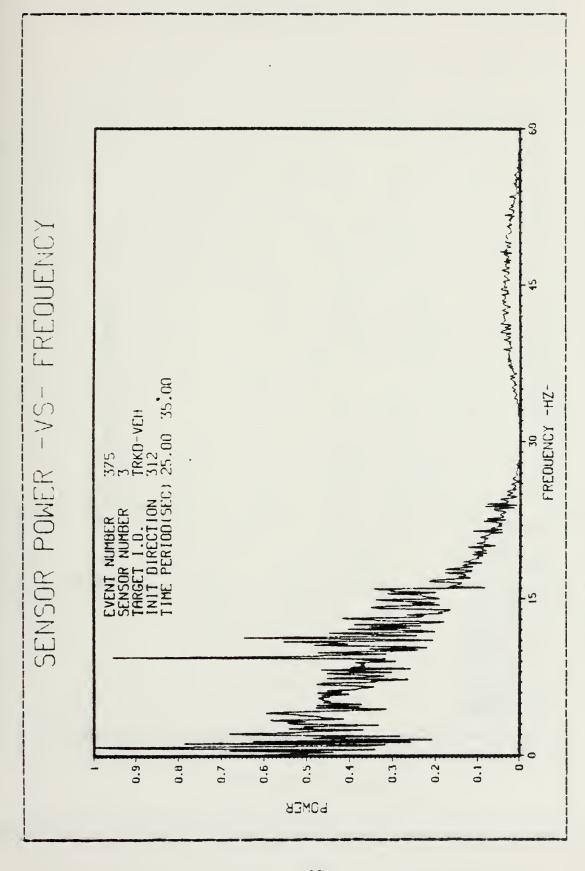
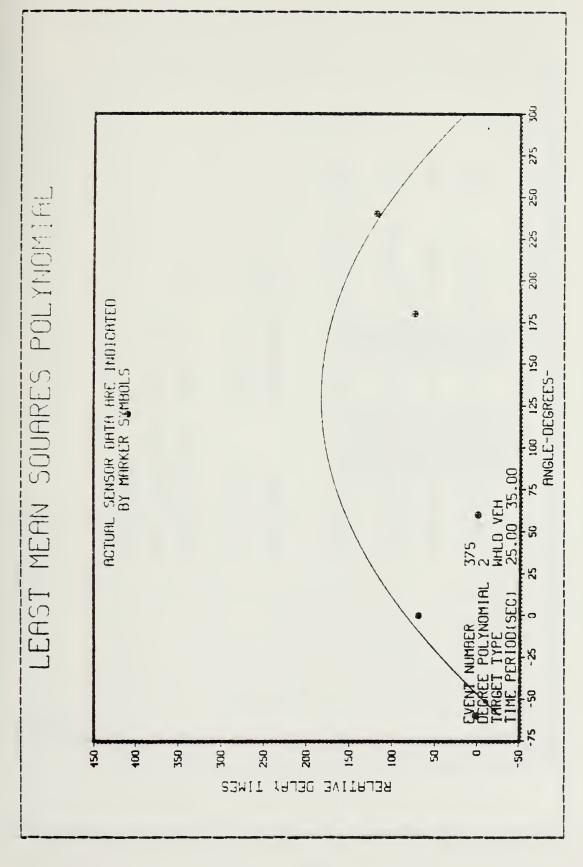


Figure 6.37 Frequency Response for Event 375





LMSP Matched Filter Direction for Event 375 Figure 6.38

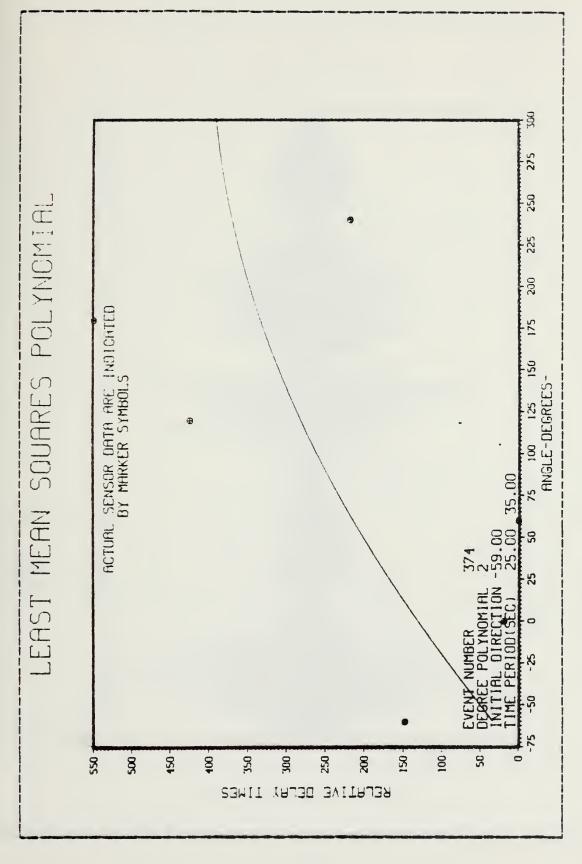


- MATCHED FILTER OUTPUT MULTIPLE TARGET

			0.00	0.00	0.00	0.00	
EVENT NUMBER 375 TIME PERIOD(SEC) 25.00 35.00	TRACKED VEHICLE DIRECTION - 0.00	SHELL BLAST DIRECTION - 312.00 HELICOPTER . DIRECTION - 0.00	SIMULATED TRKD VEHICLE TARGET FREGUENCY AMPLITUDE 0.0000	SIMULATED WHLD VEHICLE TARGET FREQUENCY AMPLITUDE 0.0000	SIMULATED HELICOPTER TARGET FREQUENCY AMPLITUDE 0.0000	SIMULATED PERSONNEL TARGET FREQUENCY AMPLITUDE 0.0000 DIRECTION 0.0000	

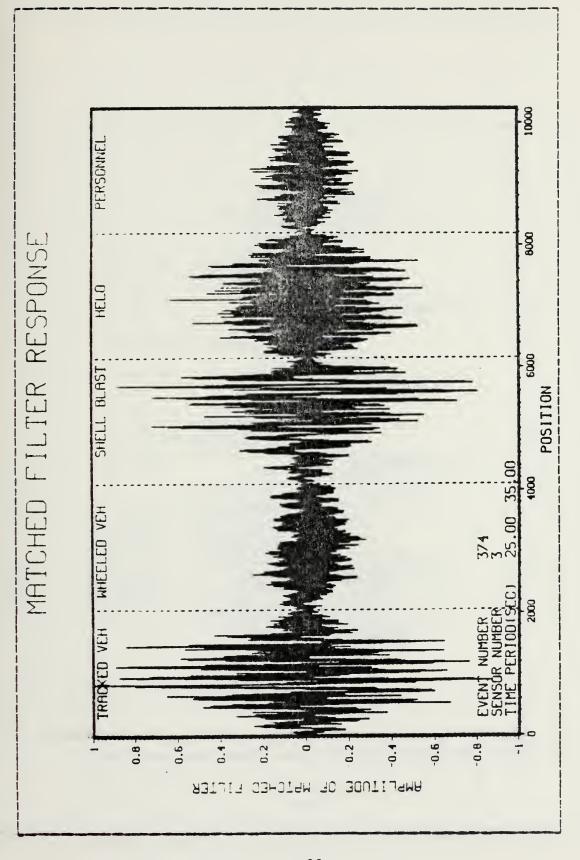
LMSP Multiple Target Direction Summary for Event 375 Figure 6.39





LMSP Initial Direction for Event 374 Figure 6.40





Matched Filter Response for Event 374 Figure 6.41



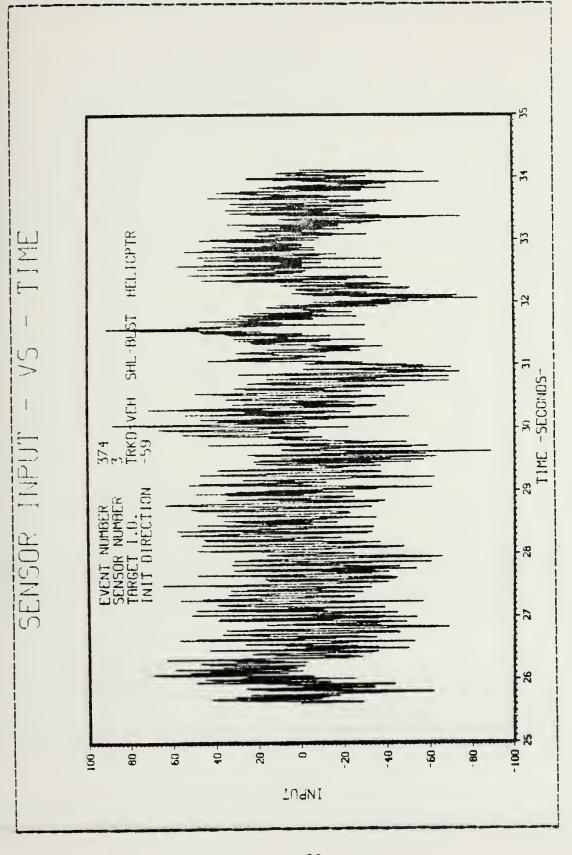
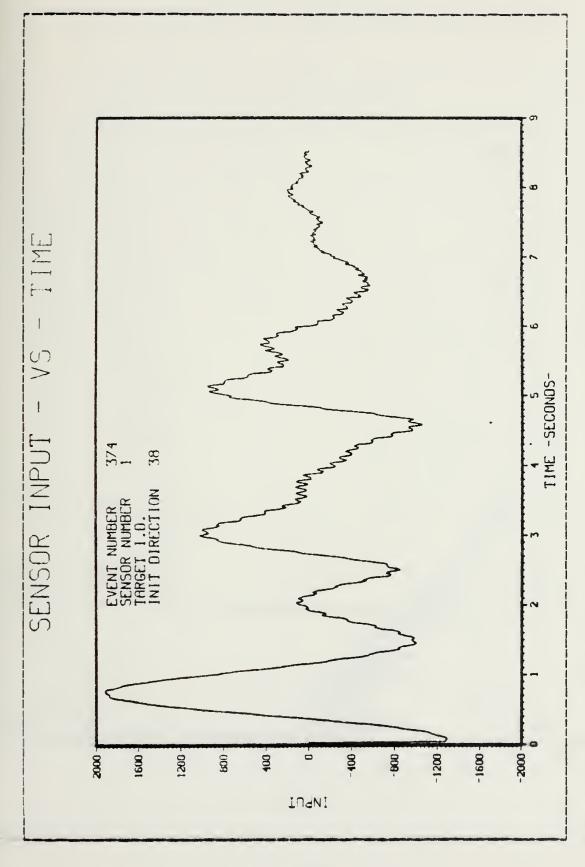


Figure 6.42 Amplitude Response for Event 374





Amplitude Response of Malfunctioning Sensor for Event 374 Figure 6.43



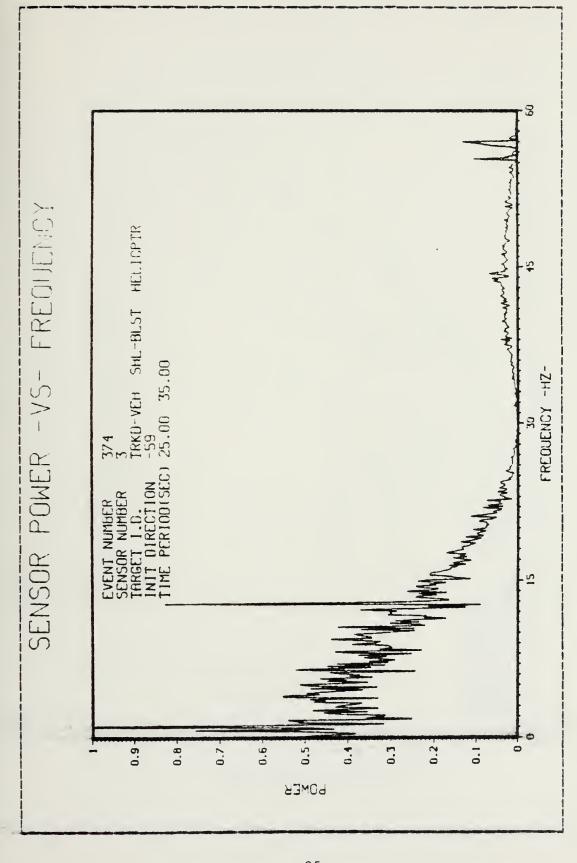
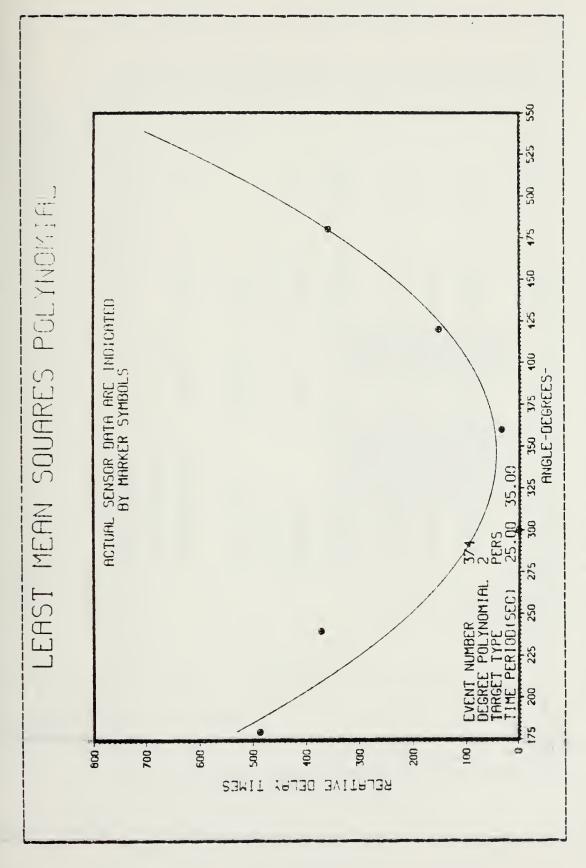


Figure 6.44 Frequency Response for Event 374





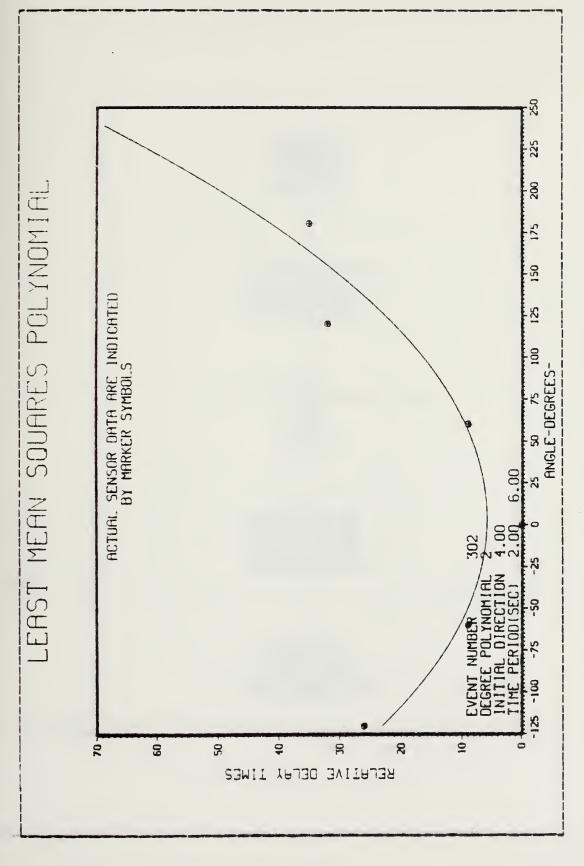
LMSP Matched Filter Direction for Event 374 Figure 6.45



- MATCHED FILTER DUTPUT MULTIPLE TARGET

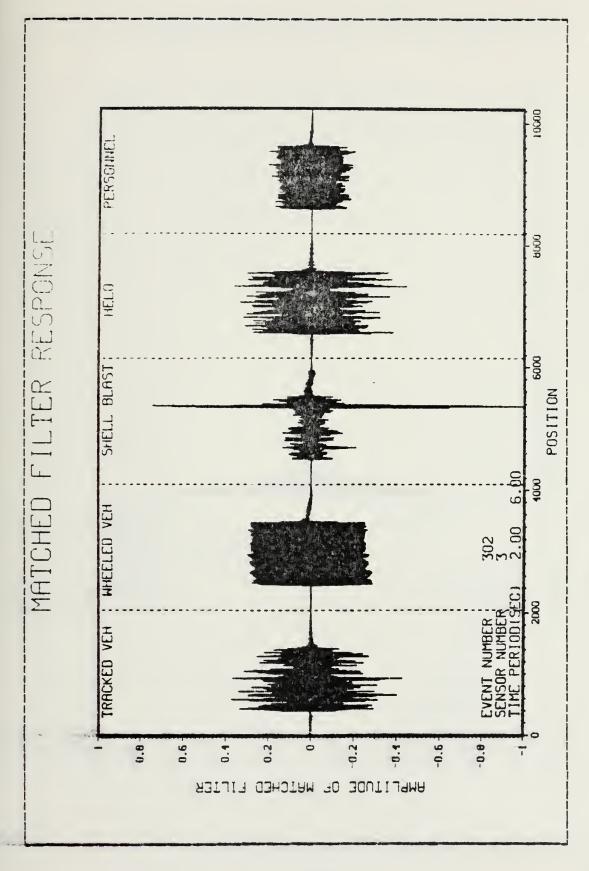
LMSP Multiple Target Direction Summary for Event 374 Figure 6.46





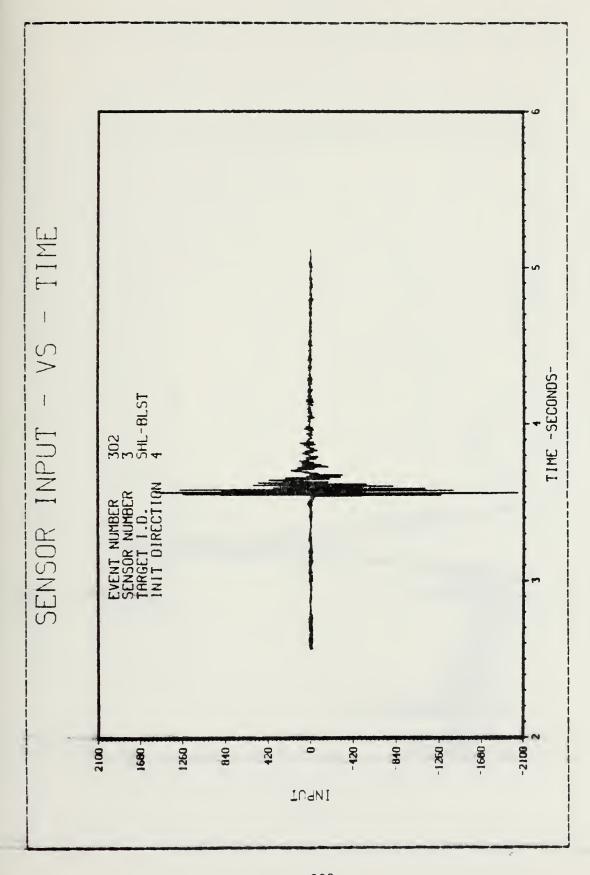
LMSP Initial Direction for Event 302 Pigure 6.47





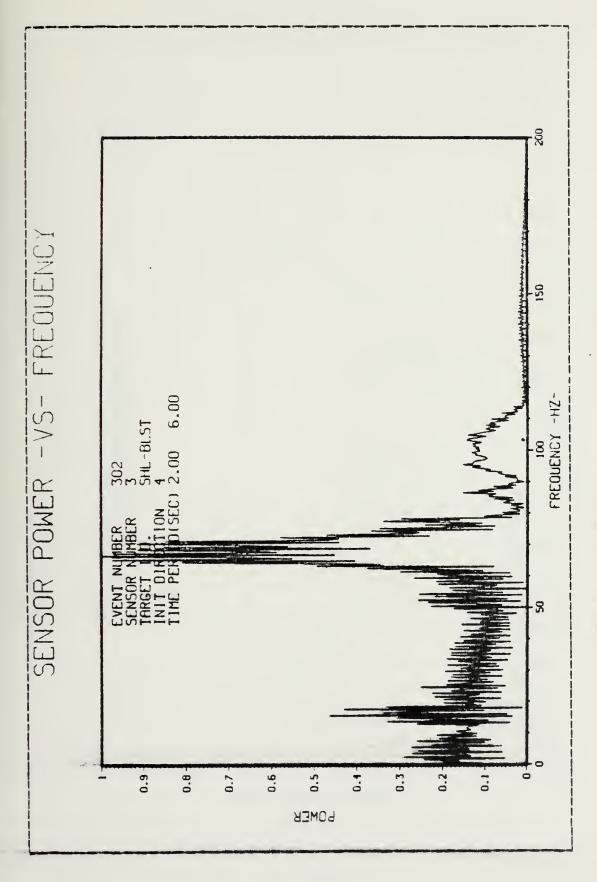
302 Matched Filter Response for Event Figure 6.48





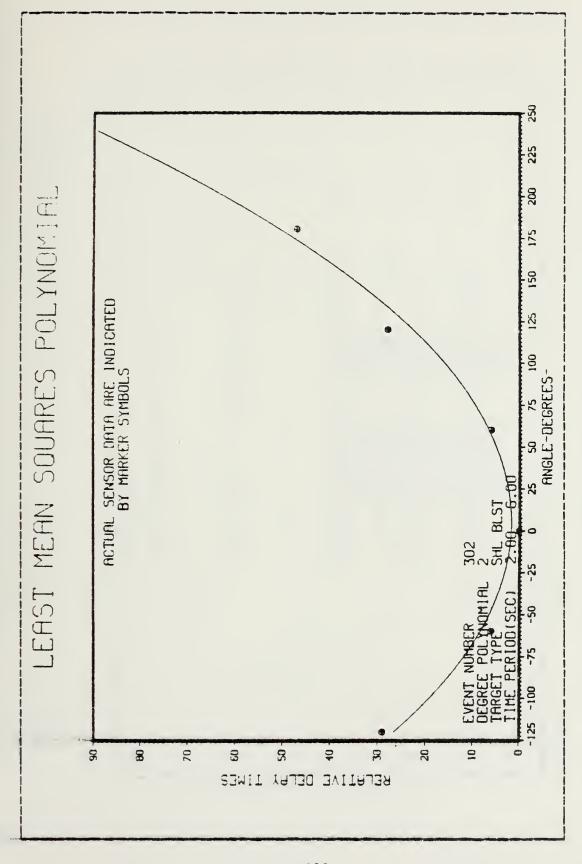
Pigure 6.49 Amplitude Response for Event 302





Pigure 6.50 Preguency Response for Event 302





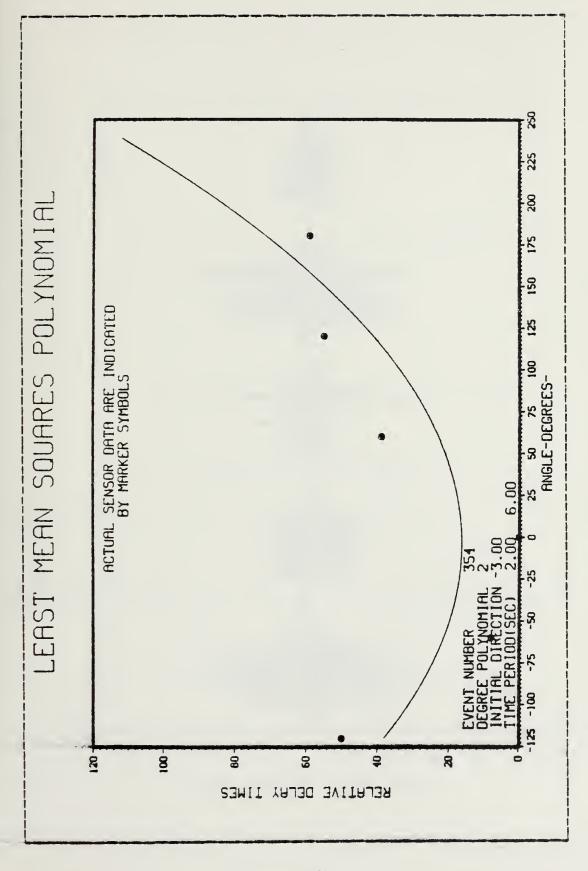
LMSP Matched Filter Direction for Event 302 Figure 6.51



MATCHED FILTER OUTPUT 0.00 0.00 0.00 0.00 O.0000 0.0000 0.0000 0.0000 0.0000 1.0000 0.0000 TARGET FREQUENCY 0.0000 0.0000 SIMULATED TRKD VEHICLE TARGET FREQUENCY DIRECTION - 6.00 TIME PERIODISEC: 2.00 DIRECTION SIMULATED WHLD VEHI AMPLITUDE DIRECTION SIMULATED PERSONNE SIMULATED HELICOPT DIRECTION AMPL 1TUBE EVENT NUMBER SHELL BLAST MULTIPLE TARGET

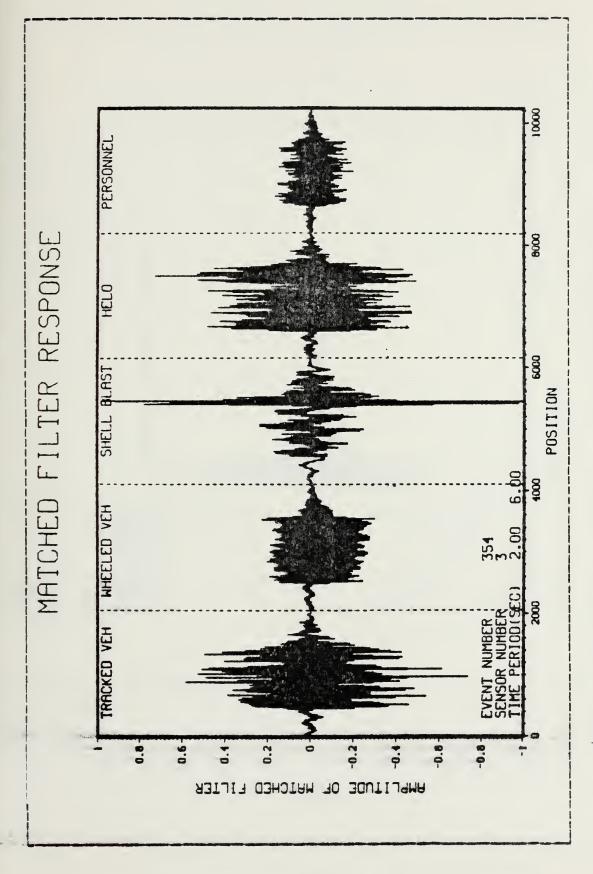
LMSP Multiple Target Direction Summary for Event 302 Figure 6.52





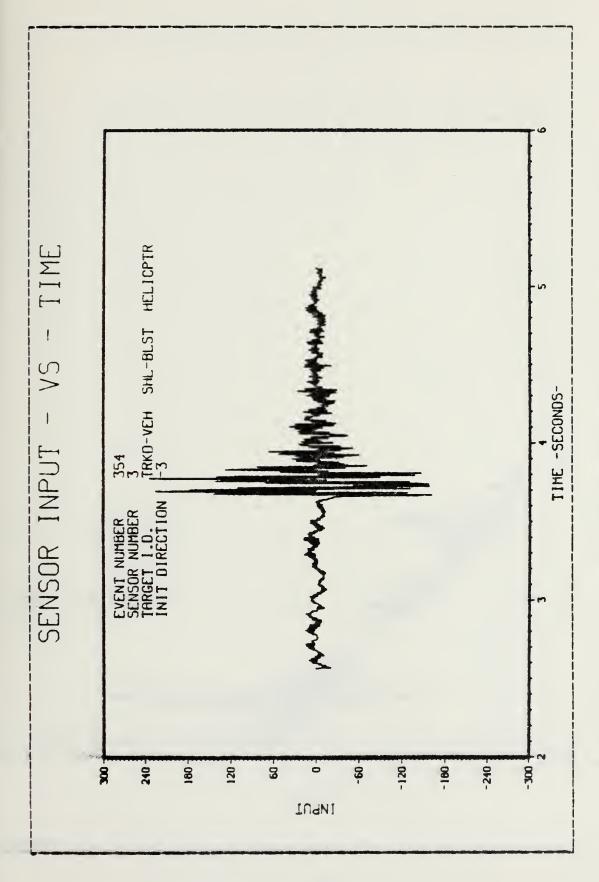
LMSP Initial Direction for Event 354 (2 - 6sec) Pigure 6.53





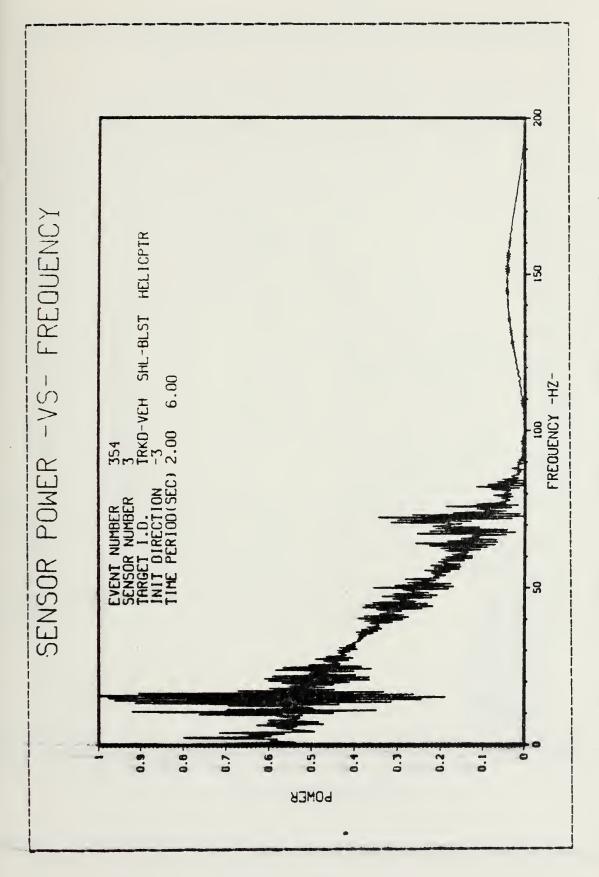
(sec) Event 354 (2 Matched Filter Response for 6.54 Figure





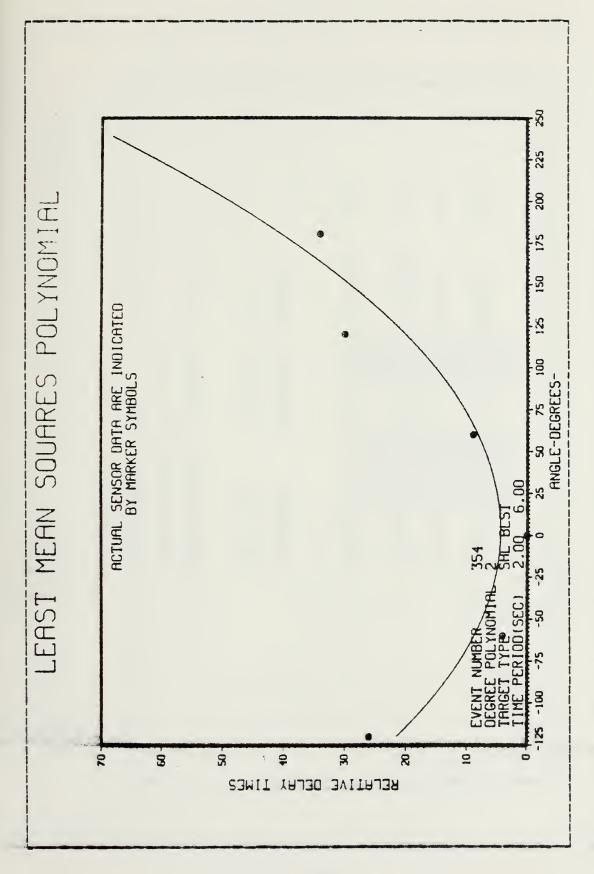
Amplitude Response for Event 354 (2 - 6sec) Figure 6.55





Prequency Response for Event 354 (2 - 6sec) Figure 6.56





LMSP Matched Filter Direction for Event 354 (2 - 6sec) Pigure 6.57

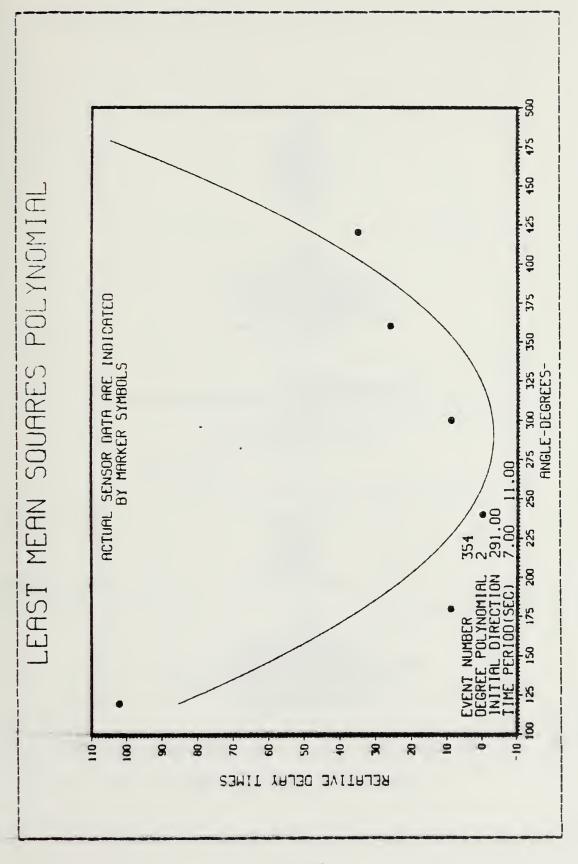


MATCHED FILTER OUTPUT 1 MULTIPLE TARGET

EVENT NUMBER 354 TIME PERIOD(SEC) 2.00 6.00 TRACKED VEHICLE DIRECTION - 332.00	SHELL BLAST DIRECTION - 3.00 HELICOPTER DIRECTION - 355.00	SIMULATED TRKD VEHICLE TARGET FREQUENCY 0.00 AMPLITUDE 0.0000 SIMULATED WHLD VEHICLE TARGET FREQUENCY 0.00 AMPLITUDE 0.0000 DIRECTION 0.0000 SIMULATED HELICOPTER TARGET FREQUENCY 0.00 BIRECTION 0.0000 SIMULATED PERSONNEL TARGET FREQUENCY 0.00 BIRECTION 0.0000 DIRECTION 0.0000	

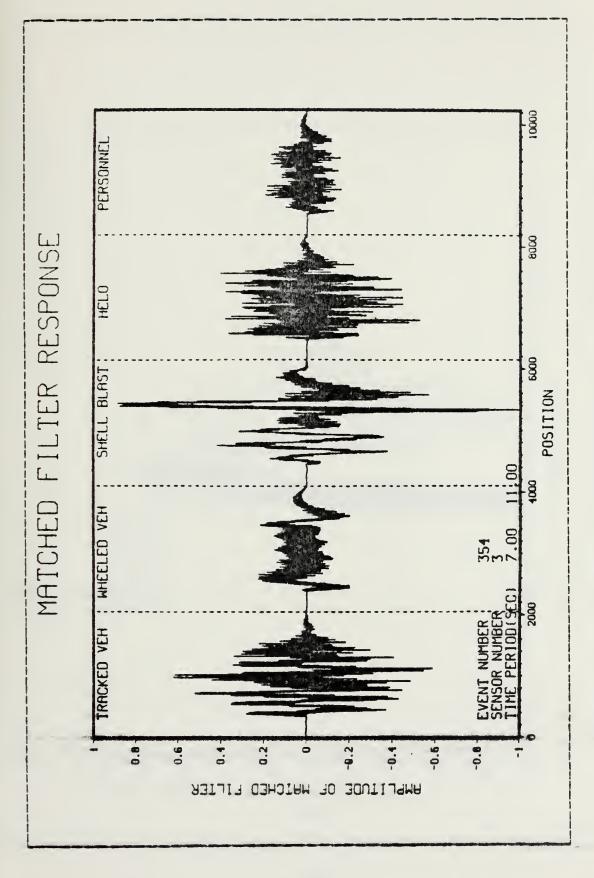
LMSP Multiple Target Direction Summary Event 354 (2 - 6sec) Figure 6.58





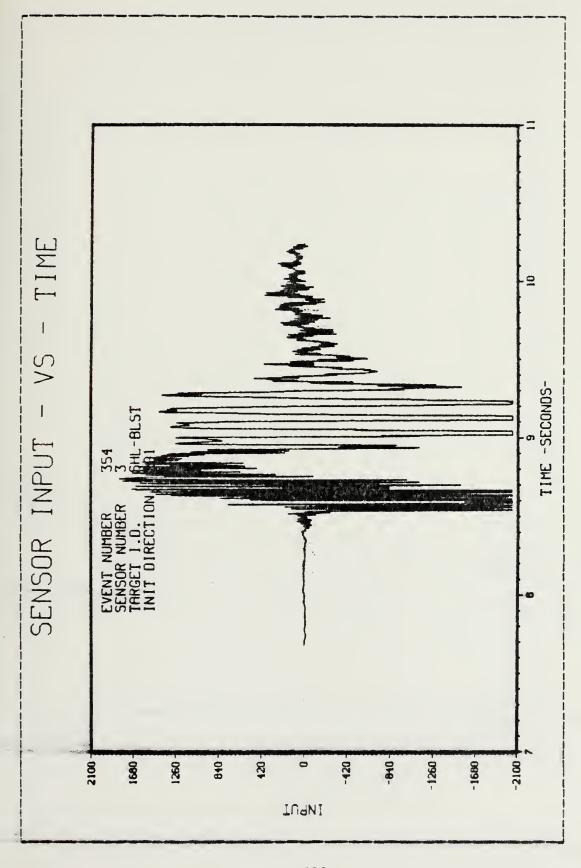
- 11sec) LMSP Initial Direction for Event 354 Figure 6.59





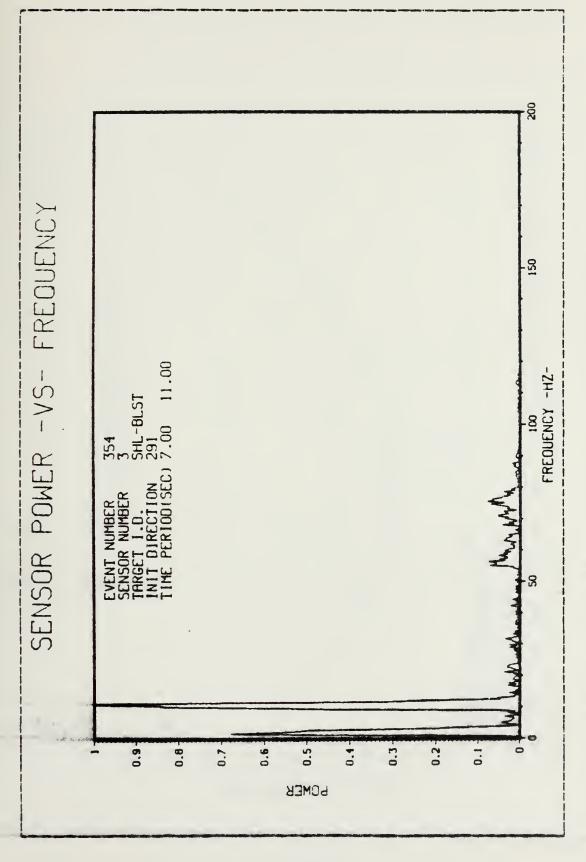
- 11sec) Matched Filter Response for Event 354 Figure 6.60





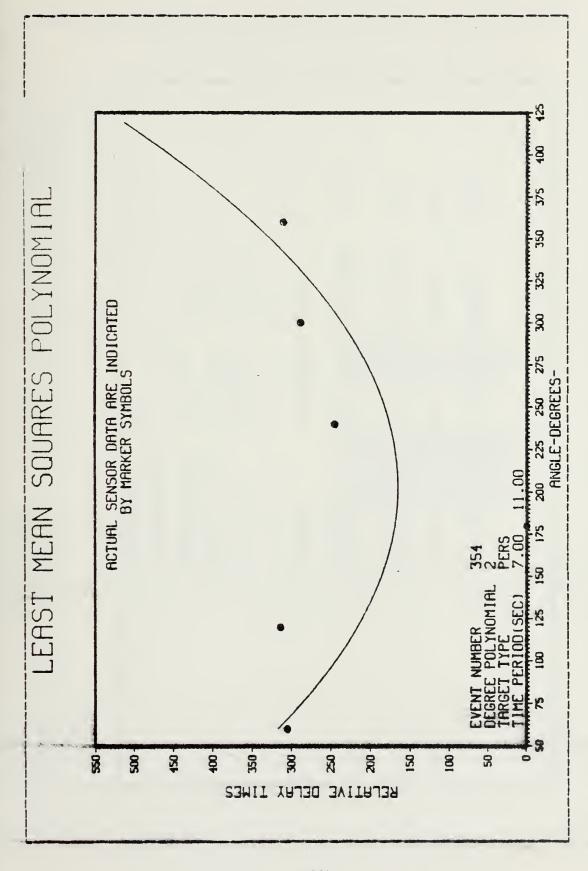
Amplitude Response for Event 354 (7 - 11sec) Figure 6.61





(7 - 11sec)Event 354 Frequency Response for Pigure 6.62





LMSP Matched Filter Direction for Event 354 (7 - 11sec) Figure 6.63



- MATCHED FILTER OUTPUT MULTIPLE TARGET

EVENT NUMBER	354	
TIME PERIOD(SEC) 7.00 11.00	7.00 11.00	
TRACKED VEHICLE	DIRECTION - 291.00	
WHEELED VEHICLE	DIRECTION - 291.00	
SHELL BLAST	DIRECTION - 291.00	
PERSONNEL	DIRECTION - 204.00	
SIMULATED TRKD VEH	SIMULATED TRKD VEHICLE TARGET FREDUENCY AMPLITUDE 0.0000	0.00
DIRECTIO SIMULATED WHLD VI AMPLITUD	101	0.00
DIRECTIO SIMULATED HELICO AMPLITUD	N 0.0000 TER TARGET FREQUENCY E 0.0000	0.00
DIRECTION SIMULATED PERSONNE AMPLITUDE DIRECTION		0.00
4		

LMSP Multiple Target Summary Event 354 (7 - 11sec) Pigure 6.64



VII. CONCLUSIONS AND RECOMMENDATIONS

The ability of the digital matched filter to detect and correctly identify actual discrete single target types was excellent. The adaptive enhancement of the matched filter scheme was found to sharpen the matched filter responses. The complications of multiple targets and continuous targets proved to be less successful. Lack of high signal to noise ratio sample signals for use as filters reduced the abiltiy of the filters to match the signals. As would be expected, the simulated target identification and direction finding operations met with success for both single and multiple targets.

The simulated data validated the usefulness of the least mean squares curve fitting method for target direction This algorithm was noted to be useful in both the relative peak amplitude response method for recoil/blast targets and the matched filter peak position method for all The highest accurracies were found using second This was due to the reduced degree polynomials. sensitivity of lower degree polynomials. Conversly, significant errors were found in the directions determined by the phase difference algorithm. These errors were possibly due to round-off error sensitivity in the software/hardware Experimental data could not be used to implementation. effectively crosscheck this finding since most of the experimental data targets were at zero degrees and the phase difference routine seemed to seek zero degrees.

A significant result was the accuracy of the blast/ recoil target direction found using only the peak amplitude responses. Directions could be found using only the relative amplitude peak positions for the array sensors and the



least mean squares curve fitting routine. This result indicated a possible counter-fire application using a greatly simplified system. This finding is felt to be significant since artillery type targets are the highest priority target type. The discrete nature of the blast/recoil seismic signals would also allow for ready separation of even a large number of combined hostile and friendly signal sources. This would allow for observerless adjustment of fire onto hostile targets. Artillery and mortar targets may, infact, be the only target types detectable at the ranges specified for a long range seismic system.

It is recommeded that further study be made in to the possible implementation of the least mean square curve fitting of the peak sensor amplitudes responses in a counter-fire system. Digital matched filters for seismic target identification and direction finding may also prove effective after further experimentation with optimum sample signal filters for the various target types. Additionally, the matched filter response may possibly be enhanced by preprocessing the seismic signals through adaptive noise cancellors.

Acoustic vice seismic matched filtering with directional microphones may be useful in target identification. Once the target has been identified, a matched filter/least mean squares based direction finding scheme could then be attempted.



APPENDIX A USERS MANUAL

The software developed provided for interactive program operation. However, further information must be provided for an initial system setup and correct program operation. To begin with, the seismic data must be transferred from magnetic tape to the IBM 3033's Mass Storage System (MSS). The data must then be transferred to the DISSPLA user's disk for analysis of the program. These data transfers may be accomplished using Jcb Control Language (JCL) procedures.

The magnetic tape volume must first be scanned to determine the storage format of its files. The JCL procedure TSCAN provides this information. A sample TSCAN job follows:

```
//JLJV1677 JOB (3026,0304), 'SMC-1677 JOHNSTON', CLASS=F

// EXEC TSCAN, VOLIN=PARK1, DCBIN='DEN=2', UNITIN='3400-4'

// EXEC TSCAN, VOLIN=PARK2, DCBIN='DEN=2', UNITIN='3400-4'

// EXEC TSCAN, VOLIN=PARK3, DCBIN='DEN=2', UNITIN='3400-4'

// EXEC TSCAN, VOLIN=PARK4, DCBIN='DEN=2', UNITIN='3400-4'
```

Once the tape scan is completed, the tape files and comments can be transferred to the MSS. Prior to this transfer however, space in the MSS must be made for these files. The procedure IEFBR14 is used for this purpose. A sample job follows:

```
//JLJTM74A JOB (3026,0304), 'SMC1677 JOHNSTON', CLASS=A
//*MAIN ORG=NPGVM1.0131P

// EXEC PGM=IEFBR14

//DD1 DD UNIT=3330V, MSVGP=PUB4B, DISP=(NEW, CATLG),

SPACE=(CYL, (4,4,3)), DSN=MSS.S3026.P302
```



```
/*

// EXEC PGM=IEFBR14

//DD1 DD UNIT=3330 V, MSVGP=PUB4B, DISP=(NEW, CATLG),

SPACE=(CYL, (4,4,3)), DSN=MSS.S3026.P314

/*

// EXEC PGM=IEFBR14

//DD1 DD UNIT=3330 V, MSVGP=PUB4B, DISP=(NEW, CATLG),

SPACE=(CYL, (4,4,3)), DSN=MSS.S3026.P319

/*
```

Each event is proceeded by a comment file for that event. These comment files can be identified from the TSCAN output as a file containing only one record. The files containing nine or eighteen records are the sensor data files for the events. Files of nine records in length are events using a circular array of nine sensors designated as a type A33 array. These sensors are all vertical motionsensing geophones. The eighteen record files contain six sensor groups of three geophones. A circular array is designated type A31 and a linear array is a type A32 array. The three geophones for each group sense either radial, transverse or vertical motion. A JCL routine, using the procedure IEBGENER, transfers the event comment and sensor data. A sample IEBGENNER job follows:



```
11
            LABEL= (1, NL, IN),
11
     DCB=(RECFM=FB, LRECL=64, BLKSIZE=2048, DEN=2, OPTCD=0)
//SYSUT2 DD
             DISP=SHR, DSN=MSS. S3026. COMMENTS (COM3 19)
//SYSIN DD DUMMY
//COPY PROC FILE= , MEM=
// EXEC PGM=IEBGENER
//SYSPRINT DD SYSOUT=A
//SYSUT1 DD UNIT=3400-4, VOL = SER = PARK2, DISP = (, PASS),
       LABEL=(&FILE, NL, IN), DCB=(RECFM=F, BLKSIZE=2048, DEN=2)
//SYSUT2 DD DISP=SHR, DSN=MSS. S3026. P319 (EMEM)
//SYSIN DD DUMMY
11
    PEND
/*
// EXEC COPY, FILE = 342, MEM = SEN1
    EXEC COPY, FILE=343, MEM=SEN2
//
    EXEC COPY, FILE = 344, MEM = SEN3
//
    EXEC COPY, FILE=345, MEM=SEN4
11
    EXEC COPY.FILE=346.MEM=SEN5
11
    EXEC COPY, FILE=347, MEM=SEN6
//
    EXEC COPY, FILE = 348, MEM = SEN7
//
    EXEC COPY, FILE=349, MEM=SEN8
//
    EXEC COPY, FILE = 350, MEM = SEN9
//
    EXEC COPY.FILE=351.MEM=SEN10
11
    EXEC COPY, FILE=352, MEM=SEN11
11
11
    EXEC COPY, FILE=353, MEM=SEN12
11
    EXEC COPY, FILE=354, MEM=SEN13
    EXEC COPY, FILE= 355, MEM = SEN14
//
    EXEC COPY, FILE=356, MEM = SEN15
//
    EXEC COPY, FILE=357, MEM=SEN16
11
11.
    EXEC COPY.FILE=358.MEM=SEN17
11
    EXEC COPY.FILE=359.MEM=SEN18
// EXEC COPY, FILE=360, MEM=SEN19
/*
```



11

Transfer of the desired event data from MSS to the DISSPLA user's disk may now be performed. A batch fortran job with the appropriate FILEDEFs to denote the various geophone's data is submitted. The RSCS/NET feature is used to send the output of this routine to the user's reader. A sample fortran job for nine sensors follows:

//JLJ83026 JOB (3026,0304), 'JOHNSTON', CLASS=A
//*MAIN ORG=NPGVM1.0090P
// EXEC FORTXCG, REGION.GO=1024K
//FORT.SYSIN DD *

LOGICAL*1 INFO1(8), INFO2(8), INFO3(8), INFO4(8), INFO5(8)

LOGICAL*1 INFO6(8), INFO7(8), INFO8(8), INFO9(8)

INTEGER*2 DATA 1 (1020) , DATA 2 (1020) , DATA 3 (1020) , DATA 4 (1020)

INTEGER*2 DATA5 (1020), DATA6 (1020), DATA7 (1020), DATA8 (1020)

INTEGER*2 DATA 9 (1020), DATB 1 (4096)

C

DO 30 J=1,4

READ (1, 100) INFO1, DATA1

READ(2.100) INFO2.DATA2

READ (3, 100) INFO3, DATA3

READ (4, 100) INFO4, DATA4

READ (8, 100) INFO5, DATA5

READ (9, 100) INFO6, DATA6

READ (10,100) INFO7, DATA7

READ (11,100) INFO8, DATA8

READ (12,100) INFO9, DATA9

100 FORMAT (8A 1, 102 (10A2))

DO 10 I = 10.1020.10

WRITE (6, 101) DATA1 (I - 9), DATA1 (I - 8), DATA1 (I - 7),

1DATA1(I - 6), DATA1(I - 5), DATA1(I - 4), DATA1(I - 3),

2DATA1(I - 2), DATA1(I - 1), DATA1(I)



```
10
      CONTINUE
      DO 20 I = 10,1020,10
      WRITE (6, 101) DATA? (I - 9), DATA2 (I - 8), DATA2 (I - 7),
      1DATA2(I - 6), DATA2(I - 5), DATA2(I - 4), DATA2(I - 3),
     2DATA2(I - 2), DATA2(I - 1), DATA2(I)
20
      CONTINUE
      DO 40 I = 10.1020.10
      WRITE (6, 101) DATA3 (I - 9), DATA3 (I - 8), DATA3 (I - 7),
      1DATA3(I - 6), DATA3(I - 5), DATA3(I - 4), DATA3(I - 3),
     2DATA3(I - 2), DATA3(I - 1), DATA3(I)
40
      CONTINUE
      DO 50 I = 10,1020,10
      WRITE (6, 101) DATA4 (I - 9), DATA4 (I - 8), DATA4 (I - 7),
      1DATA4(I - 6), DATA4(I - 5), DATA4(I - 4), DATA4(I - 3),
     2DATA4(I - 2), DATA4(I - 1), DATA4(I)
50
     CONTINUE
      DO 60 I = 10.1020.10
      WRITE (6, 101) DATA5 (I - 9) . DATA5 (I - 8) . DATA5 (I - 7) .
      1DATA5(I - 6), DATA5(I - 5), DATA5(I - 4), DATA5(I - 3),
     2DATA5(I - 2), DATA5(I - 1), DATA5(I)
60
     CONTINUE
      DO 70 I = 10.1020.10
      WRITE (6, 101) DATA6 (I - 9), DATA5 (I - 8), DATA6 (I - 7),
      1DATA6 (I - 6) , DATA6 (I - 5) , DATA6 (I - 4) , DATA6 (I - 3) ,
     2DATA6 (I - 2) , DATA6 (I - 1) , DATA6 (I)
70
      CONTINUE
      DO 80 I = 10, 1020, 10
      WRITE (6, 101) DATA7 (I - 9), DATA7 (I - 8), DATA7 (I - 7),
      1DATA7(I - 6), DATA7(I - 5), DATA7(I - 4), DATA7(I - 3),
      2DATA7(I - 2), DATA7(I - 1), DATA7(I)
80
     CONTINUE
      DO 90 I = 10, 1020, 10
      WRITE (6.101) DATAS (I - 9), DATAS (I - 8), DATAS (I - 7),
      1DATA8 (I - 6) , DATA8 (I - 5) , DATA8 (I - 4) , DATA8 (I - 3) ,
     2DATA8 (I - 2) , DATA8 (I - 1) , DATA8 (I)
```



```
90
      CONTINUE
      DO 91 I = 10,1020,10
      WRITE (6, 101) DATA9 (I - 9), DATA9 (I - 8), DATA9 (I - 7),
     1DATA9(I - 6), DATA9(I - 5), DATA9(I - 4), DATA9(I - 3),
     2DATA9(I - 2), DATA9(I - 1), DATA9(I)
91
      CONTINUE
101
       FORMAT(16,16,16,16,16,16,16,16,16)
30
       CONTINUE
       STOP
       END
/*
//GO.FT01F001 DD DISP=SHR.DSN=MSS.S3026.P383(SEN1)
//GO.FT02F001 DD DISP=SHR.DSN=MSS.S3026.P383(SEN2)
//GO.FT03F001 DD DISP=SHR.DSN=MSS.S3026.P383(SEN3)
//GO.FT04F001 DD DISP=SHR, DSN=MSS.S3026.P383(SEN4)
//GO.FT08F001 DD DISP=SHR, DSN=MSS.S3026.P383(SEN5)
//GO.FT09F001 DD DISP=SHR.DSN=MSS.S3026.P383(SEN6)
//GO.FT10F001 DD DISP=SHR, DSN=MSS.S3026.P383(SEN7)
//GO.FT11F001 DD DISP=SHR.DSN=MSS.S3026.P383(SEN8)
//GO.FT12F001 DD DISP=SHR.DSN=MSS.S3026.P383(SEN9)
//GO.SYSIN DD *
/*
11
A sample job for a six sensor group array follows:
//JLJ83026 JOB (3026,0304), 'JOHNSTON', CLASS=A
//*MAIN ORG=NPGVM1.0090P
// EXEC FORTXCG, REGION. GO = 1024K
//FORT.SYSIN DD *
C
      LOGICAL* 1 INFO 1 (8) , INFO 2 (8) , INFO 3 (8) , INFO 4 (8) , INFO 5 (8)
     LOGICAL*1 INFO6(8), INFO7(8), INFO8(8), INFO9(8)
      INTEGER*2 DATA 1 (1020) , DATA 2 (1020) , DATA 3 (1020) , DATA 4 (1020)
      INTEGER*2 DATA5 (1020), DATA6 (1020), DATA7 (1020), DATA8 (1020)
      INTEGER*2 DATA9 (1020) , DATB1 (4096)
```



```
C
      DO 30 J=1.4
          READ(1, 100) INFO1, DATA1
          READ(2, 100) INFO2, DATA2
          READ (3, 100) INFO3, DATA3
          READ (4, 100) INFO4, DATA4
          READ(8, 100) INFO5, DATA5
          READ (9, 100) INFO6, DATA6
100
       FORMAT (8A 1, 102 (10A2))
       DO 10 I = 10, 1020, 10
      WRITE (6, 101) DATA1 (I - 9), DATA1 (I - 8), DATA1 (I - 7),
     1DATA1(I - 6), DATA1(I - 5), DATA1(I - 4), DATA1(I - 3),
     2DATA1(I - 2), DATA1(I - 1), DATA1(I)
10
      CONTINUE
      DO 20 I = 10,1020,10
      WRITE (6, 101) DATA2 (I - 9), DATA2 (I - 8), DATA2 (I - 7),
      1DATA2(I - 6), DATA2(I - 5), DATA2(I - 4), DATA2(I - 3),
     2DATA2(I - 2), DATA2(I - 1), DATA2(I)
20
      CONTINUE
      DO 40 I = 10,1020,10
```

DO 40 I = 10,1020,10

WRITE (6,101) DATA3 (I - 9), DATA3 (I - 8), DATA3 (I - 7),

1DATA3 (I - 6), DATA3 (I - 5), DATA3 (I - 4), DATA3 (I - 3),

2DATA3 (I - 2), DATA3 (I - 1), DATA3 (I)

DO 50 I = 10,1020,10

WRITE (6, 10 1) DATA4 (I - 9), DATA4 (I - 8), DATA4 (I - 7),

1DATA4 (I - 6), DATA4 (I - 5), DATA4 (I - 4), DATA4 (I - 3),

2DATA4 (I - 2), DATA4 (I - 1), DATA4 (I)

DO 60 I = 10,1020,10

WRITE (6,101) DATAS (I - 9), DATAS (I - 8), DATAS (I - 7),

1DATAS (I - 6), ĎATAS (I - 5), DATAS (I - 4), DATAS (I - 3),

2DATAS (I - 2), DATAS (I - 1), DATAS (I)

60 CONTINUE

CONTINUE



```
DO 70 I = 10.1020.10
      WRITE (6, 101) DATA6 (I - 9), DATA5 (I - 8), DATA6 (I - 7),
     1DATA6 (I - 6) , DATA6 (I - 5) , DATA6 (I - 4) , DATA6 (I - 3) ,
     2DATA6 (I - 2) , DATA6 (I - 1) , DATA6 (I)
70
      CONTINUE
101
       FORMAT (16, 16, 16, 16, 16, 16, 16, 16, 16)
30
       CONTINUE
       STOP
       END
/*
//GO.FT01F001 DD DISP=SHR.DSN=MSS.S3026.P350(SEN3)
//GO.FT02F001 DD DISP=SHR, DSN=MSS.S3026.P350(SEN6)
//GO.FT03F001 DD DISP=SHR.DSN=MSS.S3026.P350(SEN9)
//GO.FT04F001 DD DISP=SHR, DSN=MSS.S3026.P350 (SEN12)
//GO.FT08F001 DD DISP=SHR, DSN=MSS.S3026.P350(SEN15)
//GO.FT09F001 DD DISP=SHR, DSN=MSS.S3026.P350(SEN18)
//GO.SYSIN DD *
/*
11
```

Two files will be returned to user's reader. The first file is the listing and diagnostics file and should be purged. The second file should be named SEN DATA. SEN DATA must now be edited. Delete the first seven lines of the file and issue the command LREC 80 to set the proper file record length.

The interactive program can be run with the complete collection of files listed in the exec MATCH. The MATCH EXEC follows:

```
ETRACE OFF

FORTGI MFILTER

GLOBAL TXTLIB FORTMOD2 MOD2 EEH IMSLSP NONIMSL

FILEDEF 10 TERMINAL

FILEDEF 05 DISK SEN DATA (PERM)

FILEDEF 07 DISK COM DATA (PERM)
```



FI 4 DISK FILTER DATA (RECFM VS PERM

FILEDEF 18 DISK DISSPLA METAFILE T4 (RECFM VBS LRECL

19065 BLOCK 19069

EXEC DISSPLA MFILTER

Entry parameters, such as the number of sensors in the array and the sampling rate, can be found in the NOSC data log for the event under study. All other interactive entries are user selected options or are self explanatory. [Ref. 12]



. APPENDIX B SAMPLE INTERACTIVE PROGRAM SESSION

match

G1 COMPILER ENTERED

SOURCE ANALYZED

PROGRAM NAME = MAIN

* NO DIAGNOSTICS GENERATED

SOURCE ANALYZED

PROGRAM NAME = ANGLE

* NO DIAGNOSTICS GENERATED

SOURCE ANALYZED

PROGRAM NAME = TIMOUT

* NO DIAGNOSTICS GENERATED

SOURCE ANALYZED

PROGRAM NAME = FREOOT

* NO DIAGNOSTICS GENERATED

SOURCE ANALYZED

PROGRAM NAME = MATCH

* NO DIAGNOSTICS GENERALED

SOURCE ANALYZED

PROGRAM NAME = MYDATA

* NO DIAGNOSTICS GENERATED

SOURCE ANALYZED

PROGRAM NAME = MAXMIN

* NO DIAGNOSTICS GENERATED

SOURCE ANALYZED

PROGRAM NAME = SPCTRM

* NO DIAGNOSTICS GENERATED

SOURCE ANALYZED

PROGRAM NAME = MULTI

* NO DIAGNOSTICS GENERATED

SOURCE ANALYZED



PROGRAM NAME = MLTPLT

* NO DIAGNOSTICS GENERATED

SOURCE ANALYZED

PROGRAM NAME = RMS

* NO DIAGNOSTICS GENERATED

SOURCE ANALYZED

PROGRAM NAME = AVG

* NO DIAGNOSTICS GENERATED

SOURCE ANALYZED

PROGRAM NAME = SIMULT

* NO DIAGNOSTICS GENERATED

SOURCE ANALYZED

PROGRAM NAME = LMS

* NO DIAGNOSTICS GENERATED

SOURCE ANALYZED

PROGRAM NAME = PLT

* NO DIAGNOSTICS GENERATED

SOURCE ANALYZED

PROGRAM NAME = SOLV

* NO DIAGNOSTICS GENERATED

*STATISTICS * NO DIAGNOSTICS THIS STEP

DISK 'T' NOT ACCESSED.

B (126) R/O

C (127) R/O

E (128) R/O

... Your Fortran program is now being loaded ...
... execution will soon follow ...

EXECUTION BEGINS...

ENTER EVENT RECORDING NUMBER-13-

383

EVENT NUMBER 383

383. A33. 10 SEPT81 5KM EOD SHOT. SET B. NO DELAY. C141 CN FINAL



AT END OF TAPE. "

ENTER SAMPLE RATE IN HERTZ-REAL-

ENTER LOW LOOK ANGLE IN DEGREES-13-

ENTER HIGH LOOK ANGLE IN DEGREES-13-200

ENTER MATCH FILTER THRESHOLD. RANGE OF 0. TO 1.0

ENTER PLOT SCALING FACTOR-REAL-.85

ENTER NUMBER OF SENSORS IN RING 6 OR 9 ONLY-I1-

ENTER SENSOR NUMBER FOR DISPLAY-I1-

ENTER NOISE THRESHOLD LEVEL -REAL1000.

ENTER DATA WINDOW SIZE FOR DIRECTION FINDING-14-0400

FOR COMPRS OUTPUT ENTER -1 -. FOR TEK618 ENTER -2-

TO CREATE SIMULATED TARGETS ENTER -1- ELSE -2-

ENTER THE FOUR SIMULATION FREQUENCIES-REAL-

ENTER FREQUENCY

0

ENTER FREQUENCY



ENTER FREQUENCY

0

ENTER FREQUENCY

120.

ENTER AMPLITUDES FOR EACH FREQUENCY-REAL-

ENTER AMPLITUDE

0

ENTER AMPLITUDE

0

ENTER AMPLITUDE

0

ENTER AMPLITUDE

4000.

ENTER TARGET ANGLE FOR EACH FREQUENCY-13-

SIX SENSORS ALLOWABLE ANGLES: 0,60,120,180,240,300

FOR NINE SENSORS: 0, 40, 80, 120, 160, 200, 240, 280, 320

ENTER ANGLE

0

ENTER ANGLE

0

ENTER ANGLE

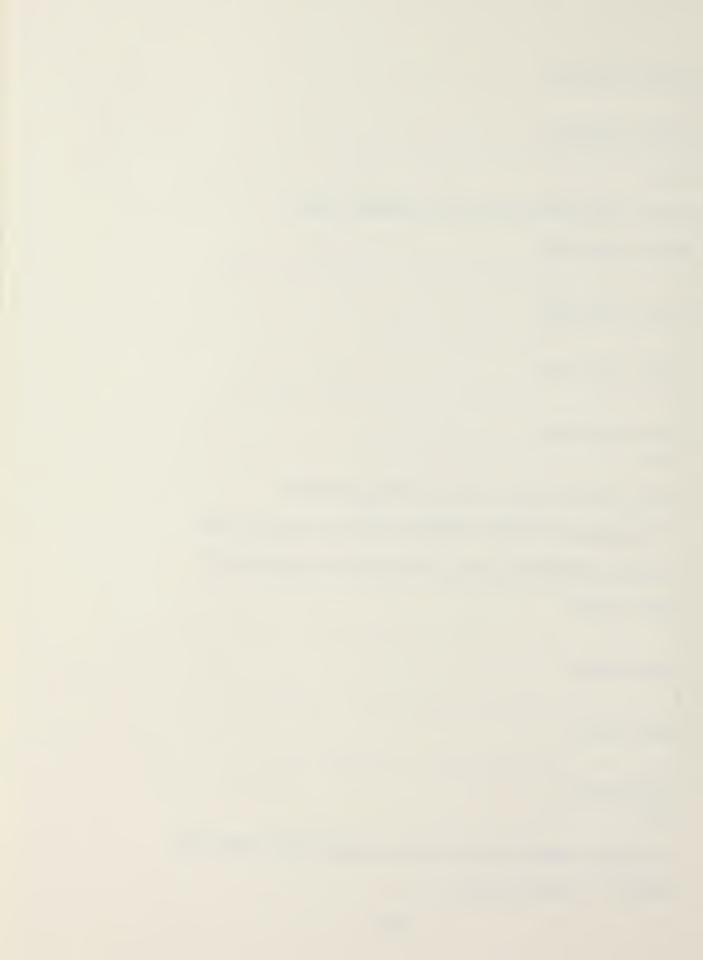
0

ENTER ANGLE

120

TO MODIFY AMPLITUDE OF SIGNAL ABOVE NOISE THRESHOLD

ENTER - 1 -, ELSE ENTER - 2 -



```
2
```

ENTER DEGREE OF POLYNOMIAL DESIRED -11-

FOR SIX SENSORS ENTER 2 - 4, FOR NINE ENTER 2 - 7

>> USING A PRE-ALLOCATED DATASET FOR UNIT FT17F001.

>> USING A PRE-ALLOCATED DATASET FOR UNIT FT18F001.

CATALOG AS TARGET? IF YES TYPE - 1 -, ELSE - 2 - 2

FOR MATCH FILTER DIREC FINDING ENTER - 1 - ELSE -2-1

FOR PHASE DELAY METHOD ENTER -1, FOR LMS ENTER -2

ENTER DEGREE OF POLYNOMIAL DESIRED -11-

FOR SIX SENSORS ENTER 2 - 4, FOR NINE ENTER 2 - 7

TO VIEW OTHER SENSORS ENTER - 1 -

ENTER - 2 - IO CONTINUE TO NEXT TIME FRAME
2



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SUEROUTINE ANGLE (DIRC,RI,R2,R3,R4,R5,R4,R7,R8,R9,NUMSEN,TESTNL,
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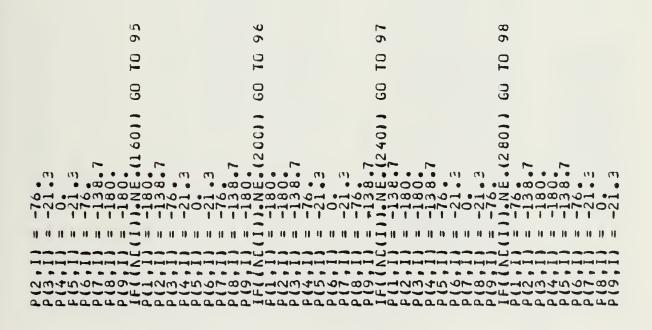
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                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                      CALL THE MATRIX SCLVER LEGTZF VIA SUBROUTINE SOLV TO FIND COEFFICIENTS FOR THE LEAST MEAN SQUARES PCLYNCMIAL
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                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                           DD 120 NR = 1, INDEX
Q = FLCAT(NR)/FLCAT(INEX)
1 + B(4)*(C**3)
2 + B(5)*(Q**4)
3 + B(5)*(Q**4)
5 + B(7)*(C**6)
6 + B(7)*(C**8)
6 + B(9)*(C**8)
7 \ NR \ = FLOAT(NR) + FLOAT(MARK - NUM)
1 \ NR \ = NR
1 \ NR \ = NR
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4 = 1, NUPSEN
= B(I7) + (X(K4)**(I7
                                                               DO 70 I7 = 2,NUMSEN

CO 6C K4 = 11,NUMSEN

CONTINUE

CO 11 NUMSEN

CO 11 NUMSEN

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EL 1 = 1,NUMS
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                                                                                                                                                                                                                                                                                                                  SCALE, TESTNO, NUMSEN, TIME, YY, VXR, BX, TIB, INCEX
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                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                              SUBROUTINE PLT (C. SCALE, TESTNO, NUMSEN, TIME, YY, VXR, BX, TTB INDE, DIRC, NFRM)
INTEGER C. TESTNC, NUMSEN, INCEX, NCR, NFRM
REAL YY (INDEX) PAKINUMSEN, INCEX, NCR, NFRM
IVXR(INCEX), TTB (NUMSEN), TIME(1024), SCALE
IF ((C) - EC (1)) CALL CGMPRS
IF ((C) - EC (1)) CALL TEKÓIB
CALL PAGE(11.0, E.5)
CALL BLChUP(SCALE)
CALL AREAZC(9-0,6.5)
CALL AREAZC(9-0,6.0)
CALL KNAME('RELATI VE DELAY TIMES', 20)
CALL KNAME('RELATI VE DELAY TIMES', 20)
CALL HEACIN('LEAST MEAN SQUARES POLYNOMIAL', 25.2.0,1)
CALL INING(TESTNO, 2.5, 65)
CALL INING(TESTNO, 2.5, 65)
CALL INING(TESTNO, 2.5, 65)
CALL INING(TESTNO, 2.5, 65)
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11 + NUM SEN
(16) * FLOA
    FMIN = F
CONTINCE
NHALF = NUMSO
DI RC = FINI
DO 77 IB = I
X (18) = X (16
CONTINCE
RETURN
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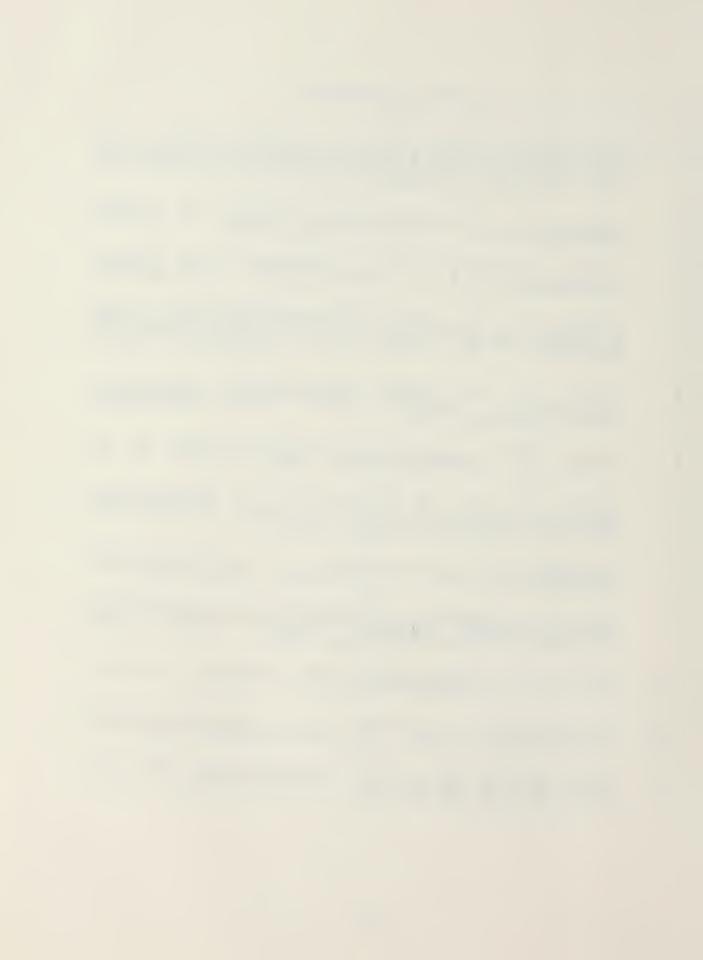
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                                                                                                                                                                         X, K, YMA X, L, YMIN,
SYNAX = _ YMAX,
SEN, K, TMAX, L, TMIN,
TNAX = _ TMAX
YMAX = TMAX
YMAX = TMAX
YMAX = TMAX
TF, YMIN = TMIN
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5, B, O, WKAREA
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CALL YAVANG COOL
CALL XIICKS (10)
CALL XIICKS (10)
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TF (TMAX)
IF ((TMAX) LE (TRIN)
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